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Large-scale coherent structures in turbulent channel flow: a detuned instability of wall-streaks

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In this paper it is shown that a modal detuned instability of periodic near-wall streaks originates a large-scale structure in the bulk of the turbulent channel flow. The effect of incoherent turbulent fluctuations is included in the linear operator by means of an eddy viscosity. The base flow is an array of periodic two-dimensional streaks, extracted from numerical simulations in small domains, superposed to the turbulent mean profile. The stability problem for a large number of periodic units is efficiently solved using the block-circulant matrix method proposed by Schmid *et al.* (*Phys. Rev. Fluids*, 2.11, 2017). For friction Reynolds numbers equal or higher than 590, it is shown that an unstable branch is present in the eigenspectra. The most unstable eigenmodes display large-scale modulations whose characteristic wavelengths are compatible with the large-scale end of the pre-multiplied velocity fluctuation spectra reported in previous computational studies. The wall-normal location of the large-wavelength near-wall peak in the spanwise spectrum of the eigenmode exhibits a power-law dependence on the friction Reynolds number, similarly to the one found in experiments of pipes and boundary layers. Lastly, the shape of the eigenmode in the streamwise-wall-normal plane is reminiscent of the *superstructures* reported in the recent experiments of Deshpande *et al.* (*J. Fluid Mech.*, 969, 2023). Therefore, there is evidence that such large-wavelength instabilities generate large-scale motions in wall-bounded turbulent flows.

Key words:

1. Introduction

Since the beginning of the XX century, pattern and coherent structure formation were studied in fluid dynamics (Lord Rayleigh 1916; Kraichnan & Chen 1989; Tuckerman *et al.* 2020). The first evidence of such structures in wall-bounded turbulent flows dates back to Kline *et al.* (1967), who found elongated velocity defects, called *streaks*, in the near-wall region of the boundary layer. The dynamics of such near-wall structures was studied by Jiménez & Moin (1991) employing numerical simulations. They showed that small computational

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domains, referred as minimal flow units, were sufficient to observe self-sustained turbulence near the wall. Moreover, the minimal flow unit concept was useful to isolate the near-wall dynamics, leading to the development of the self-sustaining wall cycle theory (Hall & Smith 1991; Hamilton *et al.* 1995). The cycle comprises a first phase during which streaks are amplified through the lift-up effect induced by streamwise vortices, a second phase during which streaks saturate and become unstable and a third phase that leads to the regeneration of streamwise vortices by nonlinear interactions, thus closing the cycle (Waleffe 1997). Butler & Farrell (1993) supported the role of the lift-up effect in the cycle with a linear transient growth computation on the turbulent channel mean profile. In addition, Jiménez & Pinelli (1999) showed that when near-wall streaks are artificially damped in the wall region, turbulence can not be sustained in the minimal flow unit. Therefore, the near-wall self-sustaining dynamics of turbulence in minimal flow units revolves around streaks and is well established.

Nevertheless, when larger domains and high Reynolds numbers are considered, the picture of wall turbulence becomes much more complex (Smits *et al.* (2011)). Kim & Adrian (1999) performed experiments on the turbulent pipe flow and reported the presence of structures having streamwise length between 12 and 14 times the pipe radius. The taxonomy of such structures in the pipe was improved by the work of Guala *et al.* (2006), who made a distinction between very-large-scale motions (VLSMs), having streamwise size between 8 and 16 pipe radii, and large-scale motions (LSMs), having streamwise length of 2-3 pipe radii. Balakumar & Adrian (2007) extended these results to turbulent channels and zero-pressure-gradient boundary layers. In these flows, the length of the large structures scales, respectively, with the channel half-height and with the boundary layer thickness. It must be noted that these experimental flows had friction Reynolds number between 500 and 2500, approximately the same range investigated in this work. Furthermore, Balakumar & Adrian (2007) observed that boundary layer flows typically have shorter VLSMs with respect to channels and pipes, a remark also made in Monty *et al.* (2007).

Simultaneously, the experimental work of Hutchins & Marusic (2007a) on turbulent boundary layers showed that, at large enough friction Reynolds number ($\gtrsim 5000$), there are two well separated peaks in the pre-multiplied streamwise fluctuation spectrum. In general, structures having a characteristic streamwise size of $O(h)$ are referred as large-scale motions (LSMs) and are associated with the region of the spectrum between the inner and the outer peak, whereas motions having a characteristic streamwise size of $O(10h)$ are those associated with the outer peak and are referred as very-large-scale motions (VLSMs) or *superstructures* (Hutchins & Marusic 2007a; Deshpande *et al.* 2023). The double peaked spectrum was observed only recently in numerical simulations of turbulent channel (Lee & Moser 2015; Hoyas *et al.* 2022), due to the high computational cost of direct numerical simulations at such high Reynolds numbers.

Investigations on large-scale and very-large-scale motions are becoming increasingly relevant for the dynamics of wall-bounded turbulent flows. These structures carry a large fraction of the turbulent kinetic energy and of the turbulent shear stress (Guala *et al.* 2006; Balakumar & Adrian 2007). Moreover, it has been realized recently that their control may be important for effective drag reduction in these flows (Marusic *et al.* 2021).

Large-scale motions extend to the wall and influence the near-wall region. Abe *et al.* (2004) reported that large-scale motions contribute to the turbulent shear stress near the wall. Likewise, Hutchins & Marusic (2007a) report the existence of a footprint of the large-scale structure near the wall. However, it must be noted that this *superposition* effect is different from the amplitude *modulation* of near-wall fluctuations reported by Hutchins & Marusic (2007b) and Mathis *et al.* (2009), which is the result of nonlinear interactions (Andreolli *et al.* 2023).

Concerning the influence of viscous effects on large-scale structures, experiments in

83 zero-pressure-gradient boundary layers (Vincenti *et al.* 2013) and turbulent pipes (Vallikivi
84 *et al.* 2015) showed that the wall-normal location of the peak at large wavelength of the
85 pre-multiplied streamwise fluctuation spectrum scales with the square root of the friction
86 Reynolds number. As argued by Hwang (2016), this means that viscous effects are non-
87 negligible even at large wavelengths.

88 The origin of large-scale structures is still uncertain. Kim & Adrian (1999), Adrian *et al.*
89 (2000) and Adrian (2007) advanced the hypothesis that such structures may be the result
90 of concatenation of smaller structures, which they identified with hairpin vortices. The
91 concatenation hypothesis has been further underpinned by Lee & Sung (2011) and Dennis &
92 Nickels (2011) in boundary layers and by Baltzer *et al.* (2013) in turbulent pipes. Furthermore,
93 Toh & Itano (2005) conjectured a co-supporting cycle where large-scale structures are
94 continuously generated by small-scale near-wall structures.

95 On the other hand, this hypothesis has been challenged by the work of Hwang & Cossu
96 (2010), who showed that large-scale structures can self-sustain even when smaller scales
97 are artificially damped. Following this study, it has been proposed that a hierarchy of self-
98 sustaining processes exists at all scales (Hwang & Cossu 2011; Cossu & Hwang 2017), each
99 resembling the near-wall cycle of Hamilton *et al.* (1995).

100 Therefore, in this view, LSMs and VLSMs are generated by some large-scale instability
101 and/or through a mechanism of transient growth on the mean shear (e.g. lift-up) as computed
102 by Del Alamo & Jimenez (2006) and Pujals *et al.* (2009) in the channel and Cossu *et al.*
103 (2009) in the boundary layer. This hypothesis is further supported by the work on the resolvent
104 analysis of McKeon & Sharma (2010) and Moarref *et al.* (2013). Whereas, secondary
105 instabilities of the large-scale structures are advocated for the transfer of energy towards
106 smaller scales (Park *et al.* 2011; Alizard 2015).

107 The existence of a bottom-up mechanism that from near-wall structures brings to large-
108 scale motions has been questioned also by Mizuno & Jiménez (2013), who showed that
109 LSMs exist even without a wall, and Zhou *et al.* (2022), who showed that the merging of
110 near-wall streaks is weakly correlated with LSMs.

111 Still, the question is not settled because there is growing statistical evidence for the
112 concatenation hypothesis (Lee *et al.* 2019; Deshpande *et al.* 2023). Interestingly, Doohan *et al.*
113 (2021) showed, using a shear stress-driven two-scale model, that energy can be transferred
114 from small scales to large scales and that this transfer corresponds to the streaks instability
115 stage. In the present study it will be shown, by means of a modal stability analysis, that a
116 detuned instability of periodic near-wall streaks can originate large-scale structures in the
117 bulk of the flow.

118 However, the application of linear stability analyses to turbulent flows is not trivial. As
119 discussed in the recent paper of Cossu (2022), two different approaches are proposed in the
120 literature. In the first, the equations are linearized around the base flow (often identified with
121 the mean flow) without including a turbulence model in the linear operator (Malkus 1956;
122 Butler & Farrell 1993; McKeon & Sharma 2010). In these studies the effect of turbulent
123 fluctuations is either neglected or included in an unknown forcing term which provides the
124 input for a resolvent analysis (McKeon 2017). In the second approach, a turbulence model
125 is included in the linear operator, often in the form of an eddy viscosity. This approach is
126 based on the work of Reynolds & Hussain (1972) and has been employed by Del Alamo
127 & Jimenez (2006); Cossu *et al.* (2009); Pujals *et al.* (2009); Park *et al.* (2011); Alizard
128 (2015); Hwang (2016) and many others. There is recent evidence of the need for modeling
129 the Reynolds stresses in linear analyses (Morra *et al.* 2019). In the case of resolvent analyses
130 on asymptotically stable mean flows, the model can be included in the forcing (Nogueira
131 *et al.* 2021). Whereas, as argued by Cossu (2022), for modal stability analyses the model
132 must be included in the linear operator. Therefore, this is the approach followed in this work.

133 Cossu (2022) showed that this method produces results consistent with direct numerical
134 simulations.

135 The rest of the paper is organized as follows. In §2 the stability problem formulation and
136 the numerical methods employed are outlined. In §3 the resulting eigenspectra and leading
137 eigenmodes are presented after a brief discussion on the considered base flow. Finally,
138 conclusions are drawn in §4.

139 2. Formulation and numerical methods

140 The objective of this work is to study the stability of turbulent near-wall streaks and its
141 connection to the appearance of large-scale structures in the flow. To this aim, Direct
142 Numerical Simulations (DNS) of turbulent channel flow are performed at different Reynolds
143 numbers (see table 1) using the `channelflow` code by Gibson *et al.* (2021). Periodic boundary
144 conditions are imposed in the streamwise and spanwise directions and no-slip conditions are
145 used at the walls ($y = 0$ and $y = 2$). The flow field is discretized by Fourier and Chebyshev
146 collocation methods in a domain having dimensions $[L_x, L_y, L_z]$.

147 Flow variables can be either nondimensionalized with respect to the channel half height h and
148 mean bulk velocity $U_b = 1/\Omega \int_{\Omega} u d\Omega$ or with respect to the friction velocity $u_{\tau} = \sqrt{\tau_w/\rho}$
149 and the viscous length scale $\delta_v = \nu/u_{\tau}$, ν being the kinematic viscosity. These quantities
150 define the turbulent Reynolds number $Re_{\tau} = u_{\tau}h/\nu$. Variables expressed in the latter (inner)
151 units are referred to with the superscript $+$, whereas, from now on, variables without any
152 superscript are scaled in the former (outer) units.

153 The size of the computational domains, the number of Fourier modes and Chebyshev
154 collocation points and the grid spacings in wall units are listed in table 1. Notice that
155 the flow cases without any subscript are those used for the base flow computation, which
156 considers rather small (although larger than the minimal flow units) domains. For instance,
157 for $Re \approx 1000$ we have used a minimal box for the logarithmic layer as in Jiménez (2013),
158 which is known to have incorrect statistics above $y \approx L_z/3 \approx 0.25h$ (Flores & Jiménez 2010;
159 Lozano-Durán & Jiménez 2014). However, it must be observed that the only purpose of
160 these simulations is to extract the near-wall coherent structures used for the stability analysis,
161 which, using spanwise-periodic conditions, allows the computation of coherent structures
162 of much longer wavelengths than those of the base flow. Whereas, the flow case indicated
163 in table 1 with the subscript L has been run with a computational domain sufficiently large
164 for allowing the development of large-scale structures, and will be used for validation of the
165 wavelengths and eigenmodes found by the stability analysis. The validity of our DNS data
166 for this purpose is further discussed in appendix A.

167 The base flow for the stability analysis is composed of the long-time averaged flow plus
168 near-wall streaky coherent structures. The mean flow profile is obtained using a semi-
169 analytical model (Reynolds & Tiederman (1967)), whereas the near-wall streaks are extracted
170 from DNS data by means of Proper Orthogonal Decomposition (POD). It has been shown in
171 previous studies (Moin & Moser (1989); Alfonsi & Primavera (2007)) that the leading POD
172 mode for the turbulent channel flow is constituted by streamwise uniform streaks. However,
173 in the near-wall region of turbulent flows, as indicated by the early works of Hamilton *et al.*
174 (1995) and more recently discussed in Jiménez (2022), due to the establishment of the wall
175 cycle, velocity streaks exhibit small inclinations with respect to the streamwise direction
176 due to the sinuous/varicose modulations induced by their secondary instability. This can be
177 visualized in figure 1, which provides a snapshot of the DNS velocity field after filtering out
178 the flow structures with spanwise wavelengths $\lambda_z^+ < 80$ and $\lambda_z^+ > 220$. The dashed lines,
179 having angle $\theta = 14^\circ$ with respect to the streamwise direction, show a mild inclination of the
180 coherent structures in the wall region. Qualitatively similar structures are found in the larger

Case	Re_b	Re_τ	L_x/h	L_z/h	N_x	N_y	N_z	Δx^+	Δz^+	Δy_w^+	Δy_c^+
C180	2800	≈ 180	π	$\pi/2$	72	129	60	≈ 7.85	≈ 4.71	≈ 0.05	≈ 4.42
C590	11000	≈ 590	$\pi/2$	$\pi/4$	96	257	96	≈ 9.65	≈ 4.83	≈ 0.04	≈ 7.24
C1000	21000	≈ 1000	$\pi/2$	$\pi/4$	200	385	200	≈ 7.85	≈ 3.93	≈ 0.03	≈ 8.2
C1000 _L	21000	≈ 1000	3π	3π	800	385	1200	≈ 11.8	≈ 7.85	≈ 0.03	≈ 8.2
C2000	45000	≈ 2000	$\pi/2$	$\pi/4$	400	633	400	≈ 8.09	≈ 4.04	≈ 0.03	≈ 10.2

Table 1: Parameters of the direct numerical simulations. Re_b and Re_τ are the Reynolds number based respectively on the bulk velocity and the friction velocity (the channel is forced with constant bulk). L_x/h and L_z/h are the domain dimension respectively in the streamwise and spanwise direction scaled with the channel half-height. N_x and N_z are the number of Fourier modes in the streamwise and spanwise direction including dealiasing. N_y is the number of Chebyshev collocation points in the wall-normal direction. Δx^+ and Δz^+ are the grid spacings in wall units in the streamwise and spanwise direction. Δy_w^+ and Δy_c^+ are the wall-normal spacings respectively near the wall and at the channel center.

181 domain DNS at $Re_\tau = 1000$ (not shown). However, it must be noted that these streaks are
 182 inclined with different angles. Whereas, we will consider a more idealized situation where
 183 the streaks are periodic along a given direction, i. e. they are all inclined with the same
 184 angle. This is a necessary approximation in order to apply the block-circulant matrix method
 185 described below.

186 For this reason, we extract by POD arbitrarily inclined coherent structures considering the
 187 Fourier decomposition of the instantaneous flow field \mathbf{u} in the streamwise (x) and spanwise
 188 (z) directions (Muralidhar *et al.* 2019):

$$189 \quad \mathbf{u}(\mathbf{x}, t) = \sum_{k_x, k_z} \hat{\mathbf{u}}_{k_x, k_z}(y, t) e^{2\pi i \left(\frac{k_x x}{L_x} + \frac{k_z z}{L_z} \right)}, \quad (2.1)$$

190 where i denotes the imaginary unit and the sum is extended to all the Fourier modes used
 191 for the discretization of the domain, while y denotes the wall-normal direction. For a given
 192 wavenumber couple $\{k_x, k_z\}$, POD is performed using the profiles $\hat{\mathbf{u}}_{k_x, k_z}(y, t)$ obtained
 193 from 2000 three-dimensional snapshots of the small domain DNS, equi-spaced in time
 194 with nondimensional time-step $\Delta t U_b/h = 0.25$. Convergence of the POD was assessed
 195 by repeating the computation with 1200 snapshots. The resulting POD eigenvalues and
 196 eigenvectors showed minor differences with respect to the case with 2000 snapshots, as
 197 the corresponding stability analysis results. Therefore, it can be concluded that the chosen
 198 number of snapshots for the POD does not qualitatively affect the outcome of the stability
 199 analysis. In one case, the POD has been repeated with 4000 snapshots, in order to address
 200 an issue described below (see discussion in section §3.1).

201 The leading POD profile obtained, $\hat{\psi}_{k_x, k_z}(y)$, is used to reconstruct a three-dimensional
 202 flow field as:

$$203 \quad \mathbf{u}_s(\mathbf{x}) = \hat{\psi}_{k_x, k_z}(y) e^{2\pi i \left(\frac{k_x x}{L_x} + \frac{k_z z}{L_z} \right)} + \text{c.c.} \quad (2.2)$$

204 (c.c.: complex conjugate). The properties of the POD (Berkooz *et al.* 1993) ensure that the
 205 mode obtained in this way is still a POD mode for the three-dimensional channel. In addition,
 206 using this method, the inclination of the resulting coherent structures can be freely chosen
 207 selecting k_x and k_z . Note that, being selected *a priori*, this inclination is not expected to
 208 vanish as the number of snapshots used for the POD is increased. Notice that, for a given
 209 couple $\{k_x, k_z\}$, there is a paired $\{k_x, -k_z\}$ mode which provides streaks inclined with the

210 same angle in the opposite direction. However, in order to obtain a two-dimensional base
 211 flow (see below), the two modes can not be employed together. It is a limitation of the current
 212 approach to consider only two-dimensional base flows.

213 The extracted structures are uniform along the direction \bar{x} , which is inclined with respect
 214 to x of an angle $\theta = \arctan [(k_x L_z)/(k_z L_x)]$. In order to make the stability analysis
 215 computationally cheaper, we exploit this spatial homogeneity by rotating the POD mode
 216 in the $y - \bar{z}$ plane, where \bar{z} is the axis perpendicular to \bar{x} . The base flow is thus constructed
 217 adding the mean turbulent flow to the rotated streaky POD mode $\overline{\mathbf{u}}_s$. Because the POD modes
 218 are defined up to an arbitrary multiplicative constant, we introduce an amplitude definition
 219 as in Alizard (2015):

$$220 \quad A = \frac{\max_{y, \bar{z}} u - \min_{y, \bar{z}} u}{2U_c}, \quad (2.3)$$

221 where u (scalar) is the streamwise component of a two-dimensional velocity field and U_c is
 222 the centerline velocity of the mean turbulent profile.

223 Before adding the streaky mode to the mean flow, the amplitude of the POD mode is
 224 normalized such that the streaks have a chosen amplitude A_s . More explicitly, let $\overline{\mathbf{u}}_s$ be the
 225 streaky mode in the rotated frame with an arbitrary amplitude (i.e. before normalization), the
 226 normalized mode $\overline{\mathbf{u}}_s^{A_s}$ is given by:

$$227 \quad \overline{\mathbf{u}}_s^{A_s} = \frac{A_s}{\frac{\max_{y, \bar{z}} \overline{\mathbf{u}}_s - \min_{y, \bar{z}} \overline{\mathbf{u}}_s}{2U_c}} \overline{\mathbf{u}}_s. \quad (2.4)$$

228 Notice that A_s is a free parameter of the problem, whose choice will be made on the basis of
 229 DNS data, as discussed in the next section. The base flow is thus constructed as:

$$230 \quad \mathbf{U}(y, \bar{z}) = \langle \mathbf{u} \rangle(y) + \overline{\mathbf{u}}_s^{A_s}(y, \bar{z}), \quad (2.5)$$

231 where $\langle \cdot \rangle$ denotes the averaging in time and in the wall-parallel spatial directions.

232 The instability of this \bar{z} -modulated base flow is thus addressed linearizing the Navier-
 233 Stokes (NS) equations around this two dimensional base flow as (Park *et al.* 2011; Alizard
 234 2015):

$$235 \quad \frac{\partial u'_i}{\partial t} = -U_j \frac{\partial u'_i}{\partial \bar{x}_j} - u'_j \frac{\partial U_i}{\partial \bar{x}_j} - \frac{\partial p'}{\partial \bar{x}_i} + \frac{\partial}{\partial \bar{x}_j} \left[(\nu + \nu_t) \left(\frac{\partial u'_i}{\partial \bar{x}_j} + \frac{\partial u'_j}{\partial \bar{x}_i} \right) \right]; \quad (2.6)$$

$$\frac{\partial u'_i}{\partial \bar{x}_i} = 0,$$

236 where $\nu = 1/Re$ is the non-dimensional viscosity and $'$ denotes the perturbations. A detailed
 237 derivation is provided in appendix B. Despite the perturbation verifies the linear equations
 238 (2.6), the base flow does not verify the corresponding steady nonlinear equations as would
 239 do in classical hydrodynamic stability analyses. This work relies on a frozen base flow
 240 assumption, which is common to numerous previous studies on the secondary instability of
 241 streaks (Schoppa & Hussain 2002; Marquillie *et al.* 2011; Park *et al.* 2011; Hack & Zaki
 242 2014; Alizard 2015; Hack & Moin 2018). This working hypothesis is a weak point of the
 243 approach because it is difficult to substantiate in a turbulent flow (see section §3.1 for further
 244 details), but it is not clear at present how to relieve it.

245 Following Park *et al.* (2011); Alizard (2015) and also the recent work of Cossu (2022),
 246 the effect of the turbulent fluctuations in the stability problem is taken into account using the

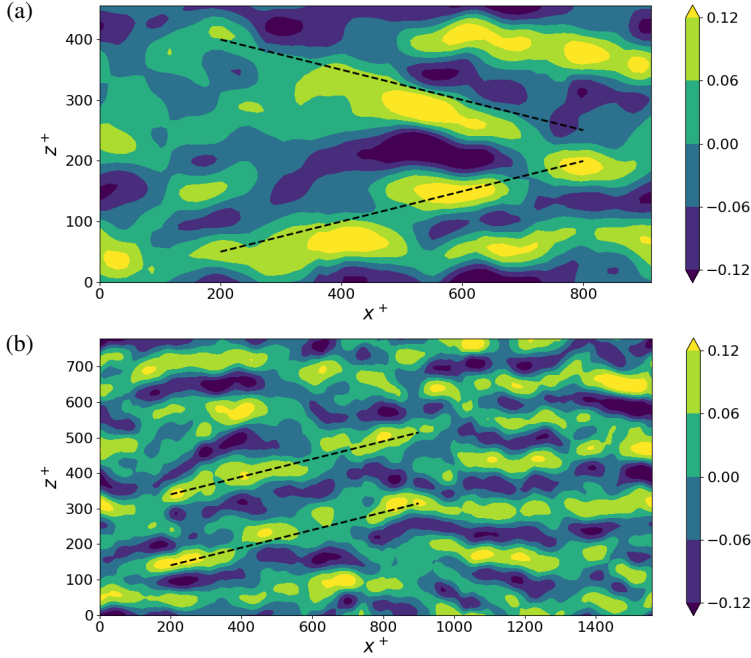


Figure 1: DNS snapshots of the streamwise velocity fluctuation at (a) $Re_\tau = 590$ and (b) $Re_\tau = 1000$ in a wall-parallel plane at $y^+ \approx 40$. The velocity field is filtered in the spanwise direction with wavelengths $80 < \lambda_z^+ < 220$. The dashed lines, having angle $\theta = 14^\circ$ with respect to the streamwise direction, show the mild inclination of the coherent structures in the wall region. These inclined streaky structures are extracted by POD for analysing their secondary instability.

247 eddy viscosity model proposed by Cess (1958):

$$248 \quad \nu_t = \frac{1}{2Re} \left\{ \left[1 + \left(\frac{Re_\tau k}{3} (2y - y^2) (3 - 4y + 2y^2) \left(1 - \frac{Re_\tau}{A} e^{|y-1|-1} \right) \right)^2 \right]^{1/2} - 1 \right\}, \quad (2.7)$$

249 with $k = 0.426$, $A = 25.4$ (Del Alamo & Jimenez 2006) and $y \in [0, 2]$. Computing the eddy
 250 viscosity from the minimal DNS data after subtraction of the projected POD mode (following
 251 Tammisola & Juniper (2016)) gave minor differences with respect to the Cess (1958) model,
 252 which do not qualitatively affect the results of the paper. This being the case, the Cess (1958)
 253 formula is preferred because it avoids the singularity at the channel center in the definition
 254 of the eddy viscosity (where the mean flow gradient tends to zero).

255 Then, a normal mode ansatz with complex frequency σ and real streamwise wavenumber
 256 α , namely,

$$257 \quad \mathbf{q}'(\bar{x}, y, \bar{z}, t) = e^{\sigma t} \tilde{\mathbf{q}}(y, \bar{z}) e^{i(\alpha \bar{x} + \sigma t)} + \text{c.c.} \quad (2.8)$$

258 is injected in the linearised NS equations for the secondary perturbation $\mathbf{q}' = (\mathbf{u}', p')^T$ and
 259 the following generalized eigenvalue problem is obtained:

$$260 \quad \sigma \mathbf{B} \tilde{\mathbf{q}} = \mathbf{A} \tilde{\mathbf{q}}. \quad (2.9)$$

261 with:

$$262 \quad \mathbf{B} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}; \quad \mathbf{A} = \begin{bmatrix} \mathcal{L} & -U_y + i\alpha v_t' & -U_{\bar{z}} & -i\alpha \\ 0 & \mathcal{L} - V_y + v_t' D_y & -V_{\bar{z}} & -D_y \\ 0 & -W_y + v_t' D_{\bar{z}} & \mathcal{L} - W_{\bar{z}} & -D_{\bar{z}} \\ i\alpha & D_y & D_{\bar{z}} & 0 \end{bmatrix};$$

$$263 \quad \mathcal{L} = -i\alpha U - V D_y - W D_{\bar{z}} + (\nu + \nu_t) \nabla^2 + v_t' D_y,$$

264 where $D_{y,\bar{z}}$ denotes differentiation with respect to y or \bar{z} , $\nabla^2 = -\alpha^2 + D_y^2 + D_{\bar{z}}^2$ is the laplacian
265 operator, $U_{y,\bar{z}}$, $V_{y,\bar{z}}$, $W_{y,\bar{z}}$ are the base flow components differentiated with respect to y or \bar{z}
266 and v_t' is the eddy viscosity differentiated with respect to y .

267 Now, note that the considered 2D base flow is periodic by construction in the \bar{z} direction.
268 Then, the instability of an array of periodic streaks can be studied efficiently using the
269 block-circulant matrix formalism proposed by Schmid *et al.* (2017). Using this method, the
270 computation of the stability of a system composed by N_u repeated sub-units is reduced to
271 N_u sub-unit-size computations, each associated to a root of unity $\rho_j = \exp(2\pi i j / N_u)$ with
272 $j = 0, 1, \dots, N_u - 1$. Here, a sub-unit is a 2D $y - \bar{z}$ domain periodic in \bar{z} .

273 The first step consists in reordering the system as a partition in N_u sub-systems corre-
274 sponding to the N_u sub-units. Thus, the eigenvalue problem is recast as:

$$275 \quad \sigma \underbrace{\begin{pmatrix} \mathbf{B} & \dots & \dots \\ \dots & \mathbf{B} & \dots \\ \vdots & \vdots & \vdots \\ \dots & \dots & \mathbf{B} \end{pmatrix}}_{\mathbf{B}'} \underbrace{\begin{pmatrix} \tilde{\mathbf{q}}^{(0)} \\ \tilde{\mathbf{q}}^{(1)} \\ \vdots \\ \tilde{\mathbf{q}}^{(N_u-1)} \end{pmatrix}}_{\tilde{\mathbf{q}}} = \underbrace{\begin{pmatrix} \mathbf{A}^{(0)} & \mathbf{A}^{(1)} & \dots & \mathbf{A}^{(N_u-1)} \\ \mathbf{A}^{(N_u-1)} & \mathbf{A}^{(0)} & \dots & \mathbf{A}^{(N_u-2)} \\ \vdots & \vdots & \dots & \vdots \\ \mathbf{A}^{(1)} & \mathbf{A}^{(2)} & \dots & \mathbf{A}^{(0)} \end{pmatrix}}_{\mathbf{A}'} \underbrace{\begin{pmatrix} \tilde{\mathbf{q}}^{(0)} \\ \tilde{\mathbf{q}}^{(1)} \\ \vdots \\ \tilde{\mathbf{q}}^{(N_u-1)} \end{pmatrix}}_{\tilde{\mathbf{q}}} \quad (2.10)$$

276 where \mathbf{A}' is the Jacobian associated to the stability problem composed of the matrices $\mathbf{A}^{(j)}$
277 (for $j = 0, \dots, N_u - 1$) describing the dynamics in a sub-unit and the interactions between
278 sub-units, and the disturbance in the j^{th} sub-unit is denoted as $\tilde{\mathbf{q}}^{(j)}$. The Jacobian matrix
279 \mathbf{A}' is block-circulant due to the specific N_u -periodic nature of the system and becomes a
280 block-diagonal matrix $\hat{\mathbf{A}}$ using the similarity transformation:

$$281 \quad \mathbf{P}^H \mathbf{A}' \mathbf{P} = \text{diag}(\hat{\mathbf{A}}^{(0)}, \hat{\mathbf{A}}^{(1)}, \dots, \hat{\mathbf{A}}^{(N_u-1)}) \equiv \hat{\mathbf{A}}, \quad (2.11)$$

282 where the transfer matrix \mathbf{P} is defined as:

$$283 \quad \mathbf{P} = \mathbf{J} \otimes \mathbf{I} \quad (2.12)$$

284 with \mathbf{J} a matrix such as $\mathbf{J}_{j+1,k+1} = \rho_j^k / \sqrt{N_u}$ for $j, k = 0, \dots, N_u - 1$ and $\rho_j = \exp(2i\pi j / N_u)$
285 the j^{th} of the N_u roots of unity. The symbol \otimes denotes the Kronecker product and \mathbf{I} the
286 identity matrix. Using this similarity transformation, the full linear stability problem is thus
287 reduced to that of the N_u sub-systems characterised by the matrices $\hat{\mathbf{A}}^{(j)}$. From equation
288 (2.11) follows that:

$$289 \quad \hat{\mathbf{A}}^{(j)} = \sum_{k=0}^{N_u-1} \rho_j^k \mathbf{A}^{(k)}. \quad (2.13)$$

290 The full eigenspectrum of the matrix \mathbf{A}' is found merging the N_u spectra of $\hat{\mathbf{A}}^{(j)}$ for
291 $j = 0, \dots, N_u - 1$. Similarly, provided \mathbf{v}_j is an eigenvector of $\hat{\mathbf{A}}^{(j)}$, the eigenfunctions

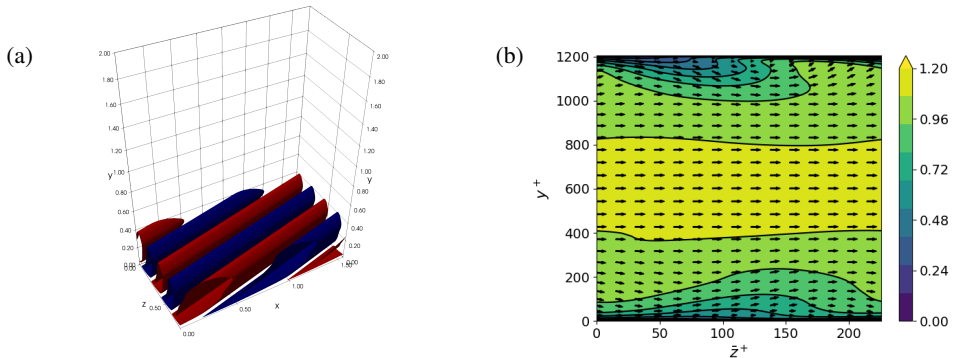


Figure 2: Base flow for the $Re_\tau = 590$ case. (a) Isocontours of the streamwise component of the POD mode with $k_x = 1$ and $k_z = 2$ ($\theta \approx 14^\circ$). (b) Base flow (POD mode plus mean flow) visualized in the $y - \bar{z}$ plane: isocontours of the streamwise component and arrows for the transverse components.

292 of the full system can be retrieved and take the form $[\mathbf{v}_j, \rho_j \mathbf{v}_j, \rho_j^2 \mathbf{v}_j, \dots, \rho_j^{N_u-1} \mathbf{v}_j]^T$ for
 293 $j = 0, \dots, N_u - 1$.

294 Physically, the argument of the root of unity $\arg(\rho_j) = 2\pi j/N_u$ ($j = 1, \dots, N_u - 1$)
 295 acts as a phase shift between the different sub-units: the farther it is from 0 (or 2π), the
 296 more desynchronised the mode is. After $2\pi/\arg(\rho_j) = N_u/j$ sub-units, the cumulative
 297 phase shift will exceed 2π , giving an estimate of the effective fundamental period of the
 298 eigenfunction of the full system. Therefore, this formalism allows the analysis of a very
 299 large-size system composed of small sub-units such as near-wall streaks, possibly prone to
 300 large-scale instabilities (Jouin *et al.* (2024)).

301 For each sub-unit, the stability problem (2.9) is discretized with $N_z = 60$ Fourier modes
 302 in the \bar{z} direction and $N_y = 129$ Chebyshev modes in the wall-normal direction. Some
 303 computations have been repeated increasing the resolution to $N_z \times N_y = 100 \times 201$, showing
 304 minor changes both on the eigenvalues (variation smaller than 2% with respect to the previous
 305 resolution) and on the shape of the eigenvectors. This validation has been carried out
 306 at the largest Reynolds number considered, which, in principle, would require the finest
 307 discretization. Hence, the coarser previously mentioned discretization is deemed appropriate
 308 and used for all the other computations.

309 3. Results

310

3.1. Base flow

311 An example of the wall-close coherent structures extracted using POD is shown in figure 2 (a).
 312 The two-dimensional base flow constructed adding these structures with amplitude $A_s = 0.25$
 313 to the mean flow is shown in figure 2 (b). As can be remarked from the arrows on this panel,
 314 the streamwise velocity modulations are subject to a nonzero mean flow component in the \bar{z}
 315 direction, due to their inclination with respect to the streamwise direction.

316 It can be noted that the extracted streaks are not symmetric with respect to the channel
 317 mid-plane. For this reason, we repeated the POD for $Re_\tau = 590$, $k_x = 1$ and $k_z = 2$ with
 318 4000 snapshots instead of 2000. Comparing panel (a) with panel (c) in figure 3 it can be
 319 observed that the asymmetry persists. Moreover, it has been remarked that the first two POD
 320 modes have comparable eigenvalues and seem to be always in phase opposition (compare
 321 panel (a) versus panel (b) and panel (c) versus panel (d)). Hence, the first two POD
 322 modes are indeed coupled. A linear combination of these two modes would be arbitrary because their

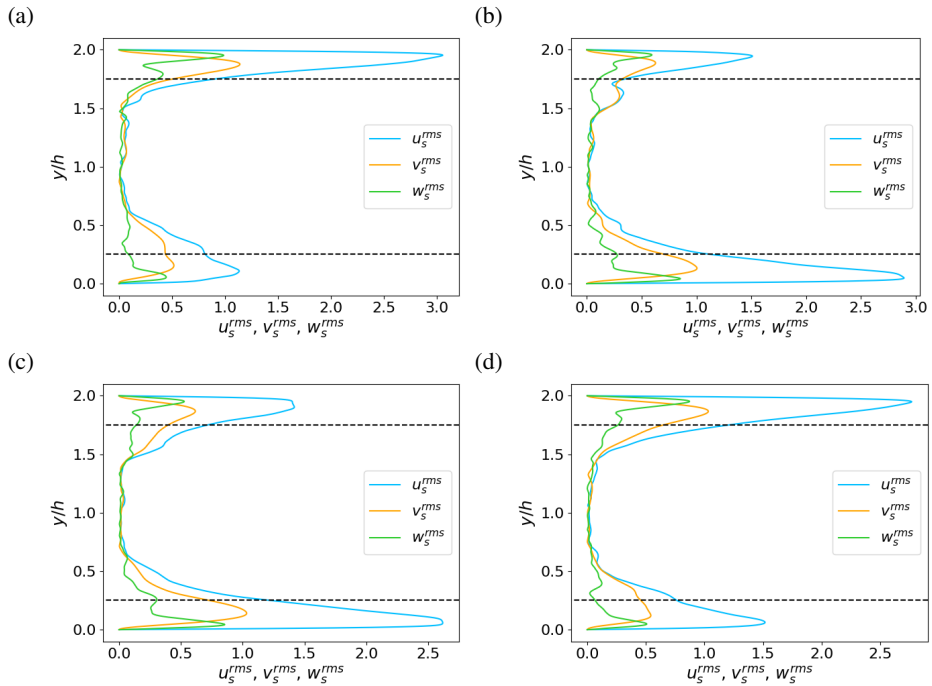


Figure 3: Root mean square of the POD streaky modes for $Re_\tau = 590$, $k_x = 1$, $k_z = 2$ (the mean is computed along x and z). The POD modes are obtained from (a,b) 2000 DNS snapshots and (c,d) 4000 DNS snapshots spaced of $\Delta t = 0.25h/U_b$ in time. (a,c) First leading mode; (b,d) second leading mode. The dashed lines denote $y/h = 0.25$ and $y/h = 1.75$ and delimit the wall regions where the turbulent flow is well resolved in the minimal domains.

323 amplitudes are not defined (we define only a relative amplitude of the streaks with respect to
 324 the mean flow, as described above). Therefore, the asymmetric mode was retained.

325 In figure 3, it can also be noted that the POD mode decreases rapidly when the distance
 326 from the wall is greater than $0.25h$, as demarcated by the dashed black lines. The flow in the
 327 minimal unit is well resolved in this region (Flores & Jiménez 2010). For larger k_x and k_z
 328 the mode is even more localized near the wall. Whereas, it is dislocated towards the channel
 329 center for $k_x = 1$ and $k_z = 1$. However, as it will be seen in the following, this particular
 330 combination of wavenumbers is not much relevant for the conclusions of the paper and is
 331 included in the results only for completeness.

332 In order to define a realistic variability range for A_s , some quantitative measurements
 333 are performed on the DNS data. For each snapshot, the equation (2.3) is applied to every
 334 $y - z$ plane and averaged in the streamwise direction and in time. The result is an amplitude
 335 of $\approx 0.3 \pm 0.01$ (blue crosses in figure 4) for all the four Reynolds number considered,
 336 where the uncertainty is given by the standard deviation in time. Evidently, this is a rough
 337 estimate because the formula (2.3) is applied to an instantaneous turbulent field, which is
 338 rather different from the base flow constituted by periodic streaky structures. Using equation
 339 (2.3), the amplitude is computed using the maximum and minimum values of the streamwise
 340 component of velocity, irrespective of the structural topology of the velocity field, in a sort
 341 of worst case scenario. Therefore, this can be considered an upper estimate on the streaks
 342 amplitude.

343 On the other hand, the root mean square (r.m.s.) peak value of the streamwise velocity

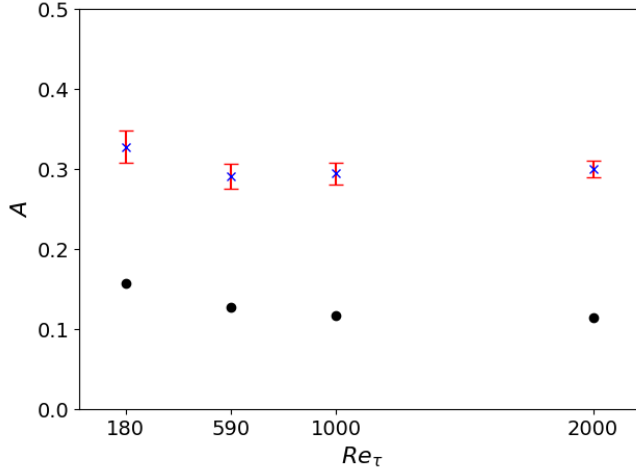


Figure 4: Amplitude A (2.3) measured in the DNS for different Reynolds numbers; the symbols denote the temporal mean value while the error bars the respective standard deviation. The circles are the peaks in the r.m.s. profiles of the streamwise fluctuation scaled with U_c (see also figure 14 (b)).

344 fluctuation (scaled with the centerline velocity for comparison) is around 0.1, as shown by
 345 the black circles in figure 4. Given the definition of r.m.s., it is sure that a greater value will
 346 be attained by the velocity fluctuation at least at some instant of time. Thus, it is argued that a
 347 realistic amplitude for the coherent structures must fall in the range $0.1 \div 0.3$. This estimation
 348 from the DNS is performed on the full velocity fluctuation, whereas the considered base
 349 streaks are made up of only one Fourier mode. The energy contained in one Fourier mode
 350 of the DNS fields is not so substantial, because of the broadband character of the turbulent
 351 velocity spectrum. Therefore, the extracted mode should not be seen as representative of the
 352 (k_x, k_z) Fourier mode alone but, rather, as an idealized representative of the whole ensemble
 353 of near-wall coherent turbulent fluctuations. In practice, the coherent turbulent spectrum was
 354 condensed in one mode. This is a very strong approximation, but it is necessary in our
 355 current approach. Note, also, that the values used by Park *et al.* (2011) and Alizard (2015)
 356 are included in the same range.

357 Moreover, the stability equations are derived assuming that the base flow \mathbf{U} verifies the
 358 following equation (see appendix B):

$$359 \quad \frac{\partial U_i}{\partial t} = -U_j \frac{\partial U_i}{\partial \bar{x}_j} - \frac{\partial P}{\partial \bar{x}_i} + \frac{\partial}{\partial \bar{x}_j} \left[(\nu + \nu_t) \left(\frac{\partial U_i}{\partial \bar{x}_j} + \frac{\partial U_j}{\partial \bar{x}_i} \right) \right] \quad (3.1)$$

360 If the right-hand side is zero, then the base flow is steady and one can legitimately formulate
 361 the stability problem as an eigenvalue problem. If \mathbf{U} is unsteady but periodic in time, one
 362 can use Floquet theory and still define eigenvalues and eigenmodes. In the case considered
 363 here, \mathbf{U} is neither steady nor periodic but has a general time dependence dictated by equation
 364 (3.1). In these cases, one usually employs the frozen base flow assumption which means that
 365 \mathbf{U} is treated as if it were steady. This implies that the right-hand side of (3.1) is somewhat
 366 neglected in the stability equations. Therefore, the assumption can be tested by comparing
 367 the time derivative of \mathbf{u}' with the time derivative of \mathbf{U} . It must be taken into account that \mathbf{u}'

368 is defined up to a multiplicative constant. Therefore, we consider

$$369 \quad \frac{1}{\|\mathbf{u}'\|} \left\| \frac{\partial \mathbf{u}'}{\partial t} \right\| \approx |\sigma| \quad \text{and} \quad \Sigma := \frac{1}{\|\mathbf{U}'\|} \left\| \frac{\partial \mathbf{U}'}{\partial t} \right\|, \quad (3.2)$$

370 where σ is the complex frequency of the mode and $\|\cdot\|$ is the L^2 -norm on the $y - \bar{z}$ domain.
 371 Then, the frozen base flow assumption is justified if the ratio $\Sigma/|\sigma|$ is small. This ratio has
 372 been computed for several of the parameter combinations explored in the following sections
 373 and has been found in most cases smaller than 1. Only in few cases, however, it is smaller
 374 than 0.1, confirming that the frozen base flow assumption is only partially corroborated. This
 375 is a limitation of the current study.

376

3.2. Leading growth rates

377 An example of eigenspectra is given in figure 5 for two of the four friction Reynolds number
 378 considered in this study, $Re_\tau = 180$ (a) and $Re_\tau = 590$ (b), for $N_u = 60$ sub-units. The
 379 colorbar shows the respective detuning factor ($\epsilon_j = j/N_u$) of the instability, which is allowed
 380 to take N_u discrete equi-spaced values in the interval $[0, 1)$. Due to the high number of
 381 sub-units considered, the eigenspectra show numerous branches. Figure 5 shows that one of
 382 these branches is almost marginally stable at $Re_\tau = 180$ and becomes prone to a detuned
 383 instability ($0.9 < \epsilon_j < 1$) at higher Reynolds number. However, it must be noted that the
 384 detuning factor depends on the reference frame, so that the effective detuning factor in the
 385 streamwise aligned ($x - z$) reference frame is lower. Nevertheless, the actual value of the
 386 detuning factor is of minor significance for the purpose of this paper. The important message
 387 of figure 5 is that an unstable branch is present at sufficiently high Reynolds number, with
 388 the leading mode corresponding to a detuned instability.

389 We verified that the number of sub-units has a weak impact both on the eigenspectra and
 390 on the eigenmodes, provided that it is sufficiently large (e.g. $N_u \geq 60$, see details in appendix
 391 D). This behaviour is expected because the number of units fixes the maximum allowable
 392 spanwise size of the modes. If this size is too small, modes linked to a large-scale spanwise
 393 modulation might not be properly resolved. On the other hand, if the domain is sufficiently
 394 large, the leading mode is properly resolved and becomes independent of the domain size.
 395 Whereas, the streaks amplitude A_s and the inclination of the base flow streaks θ strongly
 396 affect the outcome of the stability analysis, as shown in figure 6. The growth rates of the most
 397 unstable modes have similar values and trends for $Re_\tau = 590$, $Re_\tau = 1000$ and $Re_\tau = 2000$
 398 (blue, red and yellow lines), while the $Re_\tau = 180$ (green) cases show a different behaviour.

399 The dependence of the most unstable mode growth rate on the base flow streaks amplitude
 400 is displayed in figure 6 (a). As expected, the growth rate increases linearly with this parameter
 401 so that instability is found for $A_s > 0.1$ for $Re_\tau = 590, 1000, 2000$. For $Re_\tau = 180$, instability
 402 is found only at a very large amplitude $A_s > 0.25$.

403 Concerning the base flow inclination, figure 6 (b) shows that there is a peak in the growth rate
 404 at $\theta \approx 14^\circ$ for $Re_\tau = 590, 1000$ slightly displaced at $\theta \approx 7.1^\circ$ for $Re_\tau = 2000$, while the flow
 405 remains stable at $Re_\tau = 180$. This decrease of the optimal angle with the friction Reynolds
 406 number might indicate a trend towards the destabilisation of more streamwise-aligned streaky
 407 structures at higher Reynolds numbers. Whereas, for $Re_\tau = 180$ and $A_s < 0.25$, the base
 408 flow remains stable no matter the angle of the POD mode. The stability of the $Re_\tau = 180$
 409 cases can be attributed to low Reynolds number effects and is compatible with the poor
 410 evidence of large-scale structures in these flow conditions (Smits *et al.* (2011)).

411 The wavenumber along \bar{x} , namely, α , is a free parameter of the problem as well. Figure 6
 412 (c) shows that for the intermediate Reynolds number considered (590, 1000) there is a plateau
 413 in the growth rate for $\alpha \in [1.0, 2.0]$. The plateau is shifted towards smaller wavenumbers

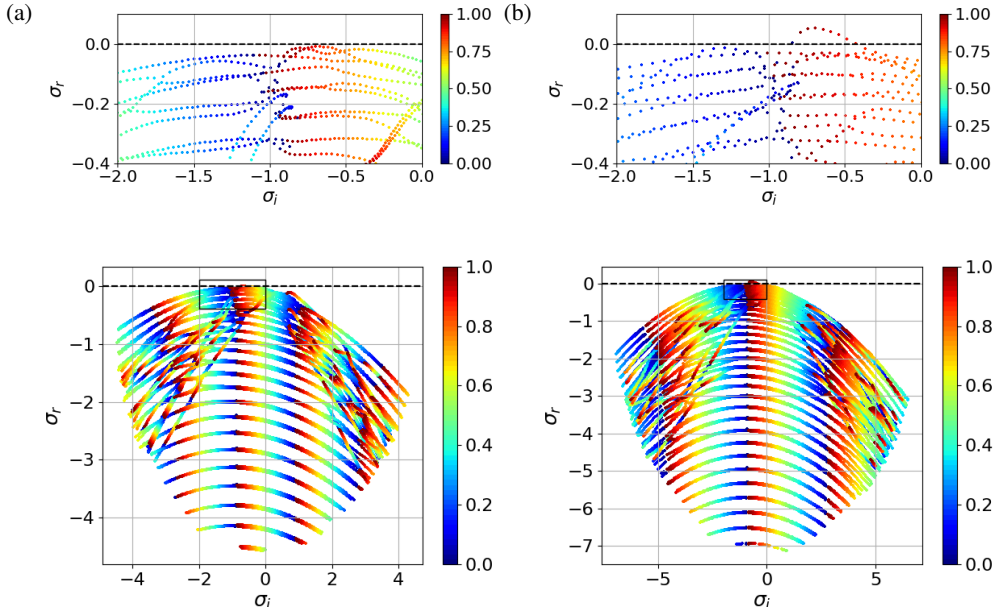


Figure 5: Eigenspectra for the $A_s = 0.25$ and $\alpha = 1.0$ case ($N_u = 60$, $\theta \approx 14^\circ$) at: (a) $Re_\tau = 180$; (b) $Re_\tau = 590$. The colors denote the detuning factor $\epsilon_j = j/N_u$ for $j = 0, \dots, N_u - 1$. The top row panels are close-ups on the portion of the spectrum denoted by solid line rectangles in the bottom row panels. The dashed black line denotes the marginal stability limit ($\sigma_r = 0$). The point with maximum growth rate σ_r defines the leading unstable mode in case of instability ($\max \sigma_r > 0$) or least stable mode in case of stability ($\max \sigma_r < 0$).

414 ($\alpha \in [0.5, 1.5]$) for $Re_\tau = 2000$. Whereas, for $Re_\tau = 180$ the growth rate is found positive
 415 for streaks amplitude $A_s = 0.2$ only for $\alpha = 2.0$.

416 Hence, the influence of α on the growth rate gives a range of unstable wavenumbers depending
 417 on the friction Reynolds number. The fact that the most unstable wavenumber decreases when
 418 Re_τ grows is consistent with the observation of coherent structures having larger streamwise
 419 wavelength at higher friction Reynolds number (compare the pre-multiplied energy spectra
 420 at different Re_τ in figure 10 and the work of Del Alamo *et al.* (2004)). As will be shown in
 421 the next section, this range of α gives a range of streamwise and spanwise wavelengths in the
 422 DNS reference frame.

423 Finally, one may ask what happens when both the base flow wavenumbers k_x, k_z are
 424 changed. To investigate this point several computations spanning $k_x = [0, 1, 2, 4, 8]$ and
 425 $k_z = [1, 2, 4, 8]$ for the two intermediate Reynolds numbers considered ($Re_\tau = [590, 1000]$)
 426 were performed having fixed $A_s = 0.20$, $\alpha = 1.0$ and $N_u = 60$. The resulting growth rates
 427 are shown in figure 7. Two aspects can be remarked: (i) the streamwise independent ($k_x = 0$)
 428 structures result always stable; (ii) increasing k_z , the maximum growth rate moves towards
 429 larger k_x . This means that the instability is found in a given range of the ratio k_z/k_x which
 430 corresponds to a range of inclination angles $\theta \approx 7^\circ \div 28^\circ$.

431

3.3. Leading eigenmodes

432 In this subsection, some observations on the leading unstable modes (or least stable modes in
 433 some cases) are presented. The $Re_\tau = 180$ case, for which evidence of large-scale motions
 434 is not compelling, will not be considered. Moreover, we fix $k_x = 1$ and $k_z = 2$ for most of
 435 the section to simplify the analysis. It will be argued that other combinations of base flow

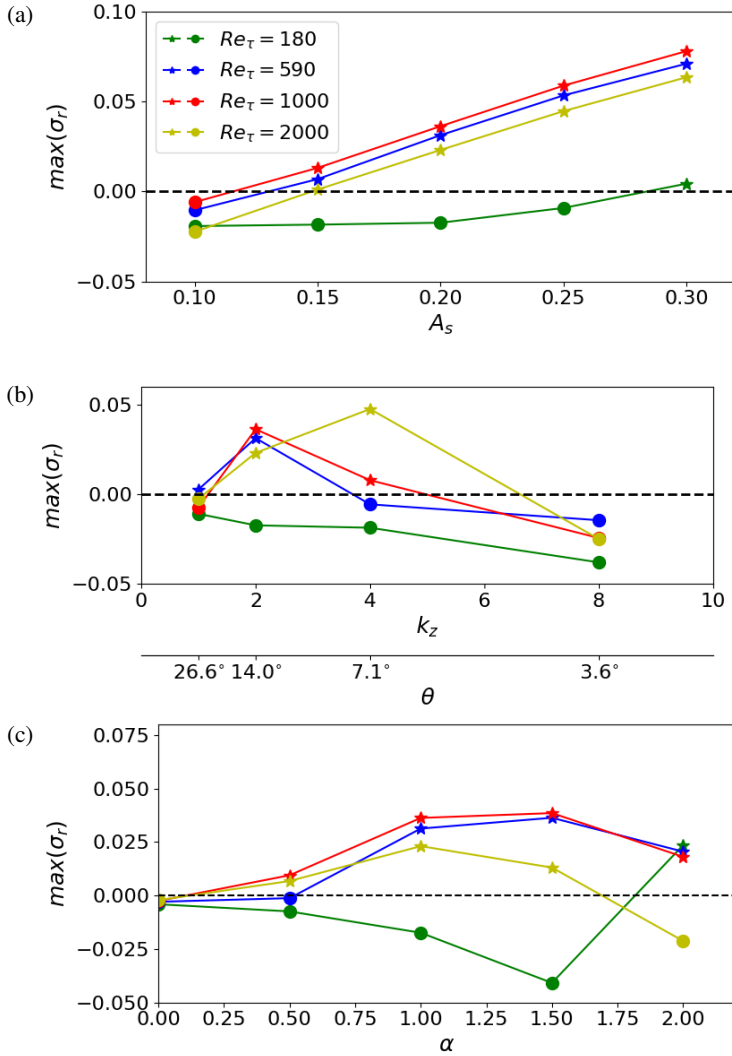


Figure 6: Growth rate of the leading unstable mode (★) or least stable mode (●) as a function of: (a) the base flow streaks amplitude (with $\alpha = 1.0$ and $k_z = 2$); (b) the inclination of the base flow (with $\alpha = 1.0$ and $A_s = 0.20$); (c) the streamwise wavenumber α (with $A_s = 0.20$ and $k_z = 2$). All computations were performed with $k_x = 1$ and $N_u = 60$.

436 wavenumbers $\{k_x, k_z\}$ lead to similar conclusions. Whereas, different $\alpha \in [0.5, 2]$ will be
 437 considered unless differently specified.

438 A closer look to the eigenmodes spatial structure is provided in figure 8, where the
 439 energy content of the mode is represented with respect to the wavelength along \bar{z} ($\bar{\lambda}_z$)
 440 and to the wall-normal position. To be more precise, consider the eigenvector $\hat{\mathbf{q}}(\mathbf{y}, \bar{z}) =$
 441 $[\hat{u}(\mathbf{y}, \bar{z}), \hat{v}(\mathbf{y}, \bar{z}), \hat{w}(\mathbf{y}, \bar{z}), \hat{p}(\mathbf{y}, \bar{z})]^T$ and its Fourier transform in the \bar{z} direction $\hat{\hat{\mathbf{q}}}$. The quantity
 442 plotted in figure 8 is $E_{uu}(\mathbf{y}, \bar{\lambda}_z) = \hat{\hat{u}}\hat{\hat{u}}^*$, where $*$ denotes the complex conjugate. It can be seen
 443 in the figure that the eigenmodes are made up of several waves with different wavelengths.
 444 Among them, one is characterized by a large wavelength and extends up to the outer region
 445 of the flow ($y/h > 0.1$). In the figure, the spanwise wavelength characterising the base flow is

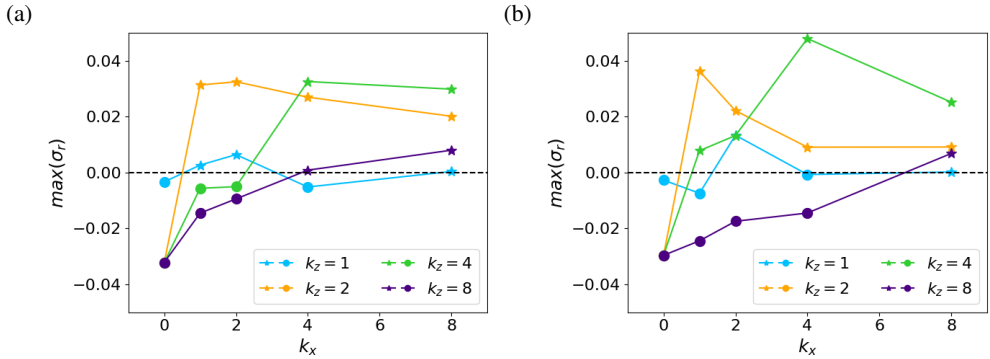


Figure 7: Growth rate of the leading unstable mode (★) or least stable mode (●) as a function of base flow wavenumbers $\{k_x, k_z\}$ for (a) $Re_\tau = 590$ and (b) $Re_\tau = 1000$. All computations were performed with $\alpha = 1.0$, $A_s = 0.20$ and $N_u = 60$.

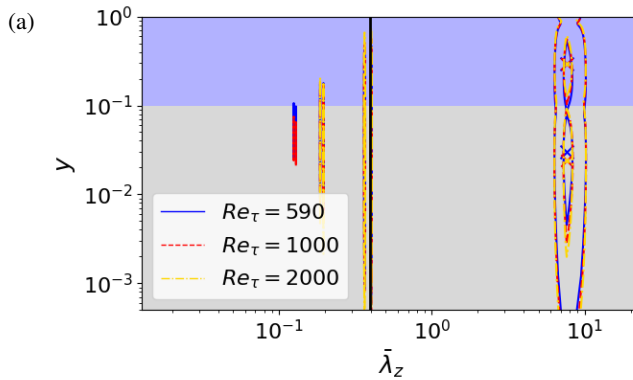


Figure 8: Spectral energy content along the \bar{z} direction of the streamwise component of the leading unstable eigenmodes (E_{uuu} in the text) for $\alpha = 1.0$ and different Reynolds numbers ($A_s = 0.20$, $k_x = 1$, $k_z = 2$). The vertical black line denotes the wavelength of the sub-units (the base flow is periodic along \bar{z} with this wavelength). The eigenmode is normalized to have kinetic energy equal to one. The contour levels are for $E_{uuu} = [10^{-4}, 10^{-1}]$. The two different background colors indicate the inner layer (light grey, for $y < 0.1$) and the bulk of the flow (light blue, for $y > 0.1$).

446 denoted by the black solid line. Therefore, it is evident that the large wavelength component
 447 of the eigenmode is much more extended than the base flow structures. We postulate that
 448 the large-wavelength modulation found in the unstable eigenmode is a large-scale structure
 449 engendered by the interaction of several base flow sub-units, i.e. several near-wall streaks.
 450 In the wall normal direction, the eigenmodes have two peaks at large wavelength: one in
 451 the wall layer ($y < 0.1$) and the other in the outer layer ($y > 0.1$). The outer one is
 452 the presumed large-scale structure, while the inner one can be interpreted as the near-wall
 453 footprint of this structure, as described by Hutchins & Marusic (2007a). The other short-
 454 wavelength modulations which are present near the wall can be attributed to the near-wall
 455 cycle (Hamilton *et al.* (1995)).

456 It is interesting to investigate how the wall-normal position of the inner peak at large
 457 wavelength depends on the Reynolds number. Experimental works in boundary layers
 458 (Vincenti *et al.* 2013) and pipes (Vallikivi *et al.* 2015) have shown that the wall-normal
 459 position of the peak of the pre-multiplied streamwise energy spectrum scales as $y_{peak}^+ \propto$

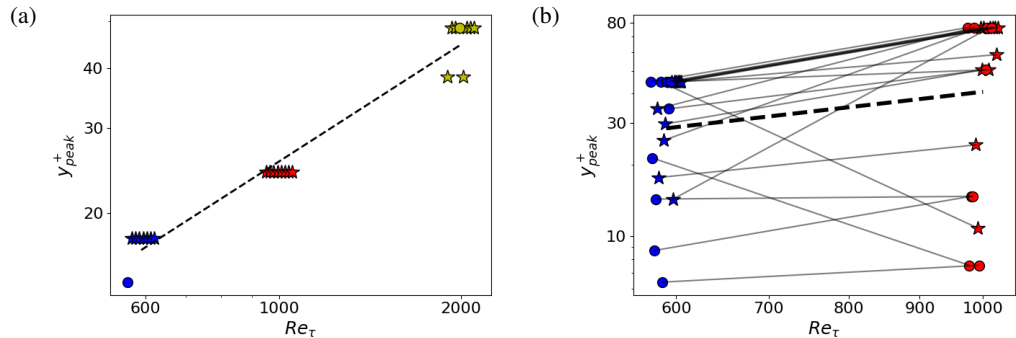


Figure 9: Wall-normal position of the inner peak as a function of Re_τ . Leading unstable modes (\star) or least stable modes (\bullet) for (a) $Re_\tau = [590, 1000, 2000]$, $\alpha = [0.5, 1, 1.5, 2]$ and $A_s = [0.20, 0.25]$ ($k_x = 1, k_z = 2, N_u = 60$) and (b) $Re_\tau = [590, 1000]$, $k_x = [0, 1, 2, 4, 8]$ and $k_z = [1, 2, 4, 8]$ ($\alpha = 1.0, A_s = 0.20, N_u = 60$). The abscissa of the symbols are jittered in order to avoid overlapping. In both (a) and (b) the black dashed line is the scaling law found by least-squares regression of all the points contained in the two plots, respectively $y_{peak}^+ \propto Re_\tau^{0.81}$ and $y_{peak}^+ \propto Re_\tau^{0.674}$. In (b) the grey lines join the points corresponding to a given $\{k_x, k_z\}$ couple from one Reynolds number to the other.

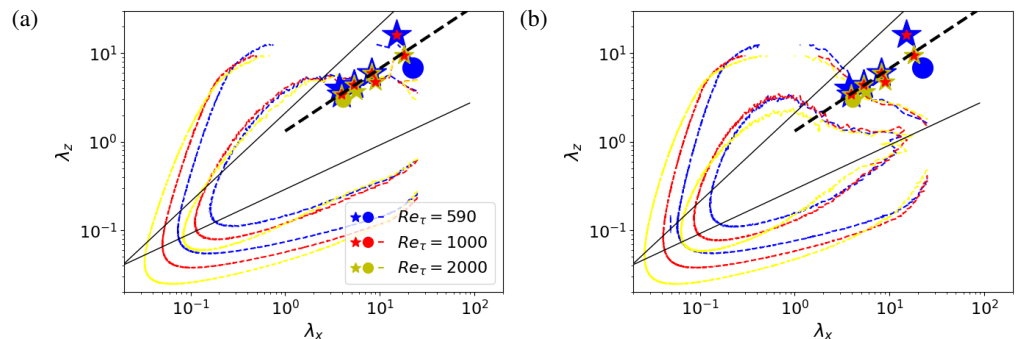


Figure 10: Wavelengths of the (a) streamwise and (b) wall-normal velocity component of the leading unstable modes (\star) or least stable modes (\bullet) reported in the DNS reference frame for several cases. The contours are the reference DNS pre-multiplied spectra at $y/h \approx 0.3$ for the (a) streamwise and (b) wall-normal velocity fluctuations by Del Alamo *et al.* (2004) and Hoyas & Jiménez (2006). The wavelengths of the eigenmodes follow the thick dashed line given by $\lambda_z = 1.32\lambda_x^{0.71}$. The solid black lines are the scaling laws $\lambda_z \propto \lambda_x$ and $\lambda_z \propto \lambda_x^{0.5}$ reported in Del Alamo *et al.* (2004).

460 $Re_\tau^{0.5}$ (or $y_{peak}/h \propto Re_\tau^{-0.5}$), meaning that even large-scale structures are affected by
 461 viscous effects (Hwang 2016). In figure 9 (a), the wall-normal position of the inner peak for
 462 the leading eigenmodes computed with $Re_\tau = [590, 1000, 2000]$, $\alpha = [0.5, 1, 1.5, 2]$ and
 463 $A_s = [0.20, 0.25]$ is plotted as a function of the friction Reynolds number. A power-law
 464 least-squares regression applied to all these points gives the scaling law $y_{peak}^+ \propto Re_\tau^{0.81}$
 465 (dashed line in the figure). This result is in between the experimental findings recalled above
 466 and the scaling $y_{peak}^+ \propto Re_\tau^{0.898}$ found by Hwang (2016) from the primary transient growth
 467 of the mean flow. Note that we considered the spectrum with respect to $\bar{\lambda}_z$ instead of the
 468 spectrum with respect to λ_x , but the location of the inner peak remains unchanged when
 469 the reference frame is rotated from \bar{z} to x . Thus, the scaling law we found can be directly
 470 compared to the previously mentioned ones.

471 To assess if the large-scale modulations observed in the leading eigenmodes can represent
 472 large-scale structures populating turbulent flows, a comparison of their wavelengths with DNS
 473 data is presented. Again, the leading eigenmodes computed with $Re_\tau = [590, 1000, 2000]$,
 474 $\alpha = [0.5, 1, 1.5, 2]$ and $A_s = [0.20, 0.25]$ are considered. For each of these, the prominent
 475 $\bar{\lambda}_z$ is extracted from the Fourier spectrum (the $\bar{\lambda}_z$ corresponding to the peaks in figure 8).
 476 Then, the streamwise and spanwise wavelengths in the DNS frame (λ_z, λ_x) must be obtained
 477 from those in the rotated frame ($\bar{\lambda}_z$ and $\bar{\lambda}_x = 2\pi/\alpha$). This can be done considering the
 478 large-scale modes as waves whose crests are inclined with respect to the \bar{z} axis of an angle
 479 $\beta = \arctan(\bar{\lambda}_x/\bar{\lambda}_z)$. The inclination with respect to the z axis is $\beta + \theta$, hence:

$$480 \quad \lambda_x = \bar{\lambda}_x \frac{\cos(\beta)}{\cos(\beta + \theta)}, \quad \lambda_z = \bar{\lambda}_z \frac{\sin(\beta)}{\sin(\beta + \theta)}, \quad (3.3)$$

481 with θ the angle of inclination of the rotated frame. These transformations are equivalent to
 482 a rotation by θ of the wavenumber-vector:

$$483 \quad \begin{pmatrix} \frac{2\pi}{\lambda_x} \\ \frac{2\pi}{\lambda_z} \end{pmatrix} = \begin{pmatrix} \cos \theta & -\sin \theta \\ \sin \theta & \cos \theta \end{pmatrix} \begin{pmatrix} \frac{2\pi}{\bar{\lambda}_x} \\ \frac{2\pi}{\bar{\lambda}_z} \end{pmatrix}. \quad (3.4)$$

484 The results for the streamwise and wall-normal velocity components are shown in figure 10
 485 along with the respective DNS pre-multiplied spectra by Del Alamo *et al.* (2004) and Hoyas
 486 & Jiménez (2006). The characteristic sizes of the unstable modes (blue/red/yellow stars) are
 487 included in the large-wavelength portion of the DNS spectra, showing that these eigenmodes
 488 may represent the large-scale modulations found in the turbulent flow. The spectra are taken
 489 at a wall-normal position $y/h \approx 0.3$, which is not far from the outer peak of the mode (see
 490 figure 8 and figure 12). The premultiplied energy spectra of the spanwise component are
 491 equivalent to the wall-normal ones, hence are not shown.

492 Particularly, the wavelengths of the eigenmodes are found to scale according to the law
 493 $\lambda_z \propto \lambda_x^\gamma$ with $\gamma \approx 0.71$ (thick dashed line in figure 10). This scaling law is comprised between
 494 the scalings reported by Del Alamo *et al.* (2004), namely $\lambda_z \propto \lambda_x^{0.5}$ and $\lambda_z \propto \lambda_x$. Del Alamo
 495 *et al.* (2004) report the laws in terms of λ_x/y and λ_z/y , whereas we are considering only the
 496 eigenmode outer peak wall-normal position, so y is fixed and included in the proportionality
 497 constant. This is because the eigenmode is representative only of one large-scale structure
 498 at a time, whereas the scaling of the DNS spectra with respect to y is due to the whole
 499 hierarchy of structures present in the flow. These two scaling laws are reported in the panels
 500 of figure 10 as solid black lines. The first one is mainly followed by the streamwise velocity
 501 fluctuation, the second characterizes the transverse fluctuations (wall-normal and spanwise).
 502 Whereas, our eigenmodes display the same prominent wavelengths both for the streamwise
 503 and for the wall-normal component (the same is true also for the spanwise component, not
 504 shown). Consistently, the scaling law is intermediate between the two given by Del Alamo
 505 *et al.* (2004).

506 Until now, only the couple $\{k_x, k_z\} = \{1, 2\}$ of base flow wavenumbers has been
 507 considered. Whereas, in figure 11 it is shown how the choice of these parameters affects
 508 the characteristic wavelengths of the eigenmodes (reported in the DNS reference frame as
 509 explained above). It can be remarked that the wavelengths are rather independent of both
 510 wavenumbers except in two cases: (i) when $k_x = 0$, but this is not much relevant because
 511 these streaks are always stable; (ii) when $k_x = 8$ and $k_z < 8$, but this case also is not much
 512 relevant because corresponds to structures inclined of an angle $\geq 45^\circ$ with respect to the
 513 streamwise direction. These large inclination angles are unlikely to be found near the wall
 514 and are included only for completeness. Also the Reynolds number dependence illustrated
 515 in figure 9 (a) is generally not affected by the variation of k_x and k_z , as it is demonstrated by

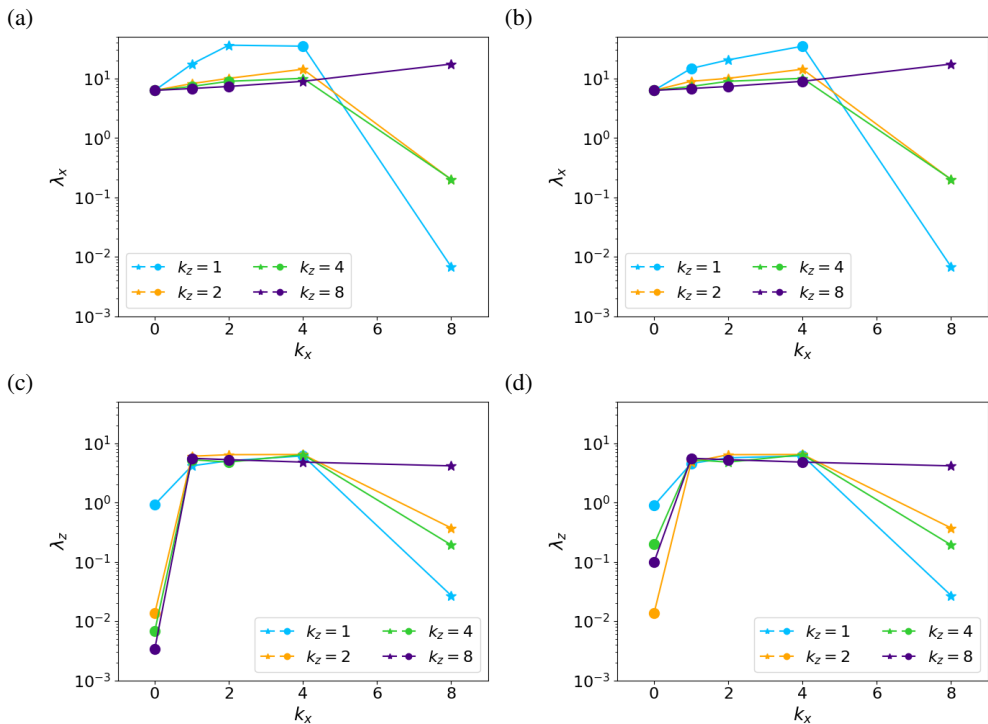


Figure 11: Streamwise (a,b) and spanwise (c,d) wavelengths of the streamwise velocity component of the leading unstable modes (\star) or least stable modes (\bullet) reported in the DNS reference frame as functions of base flow wavenumbers $\{k_x, k_z\}$ for (a,c) $Re_\tau = 590$ and (b,d) $Re_\tau = 1000$. All computations were performed with $\alpha = 1.0$, $A_s = 0.20$ and $N_u = 60$.

516 panel (b) of the same figure. It can be seen that for most of the $\{k_x, k_z\}$ couples the position
 517 of the trend increases and on average the trend is consistent with the law reported in panel (a).
 518 Therefore, it can be concluded that the main features of the eigenmodes are robust enough
 519 to allow us to focus only on a given couple $\{k_x, k_z\}$ for the rest of the paper.

520 Finally, the spatial structure of the unstable modes is presented in figure 12 (a-b-c). One of
 521 the leading unstable eigenmodes is shown in a $x - y$ plane for each $Re_\tau \in \{590, 1000, 2000\}$.
 522 It can be seen that the Reynolds number does not have a major influence on the shape
 523 of the eigenmode. This visualization recalls a conceptual sketch suggested in figure 13 of
 524 Deshpande *et al.* (2023), where an experimental boundary layer is considered. Despite the
 525 Reynolds number considered in this study is too low compared to the experiment, the large-
 526 scale modulation of the eigenmode is found to reproduce the shape of a *superstructure*. To
 527 make a direct comparison, following Deshpande *et al.* (2023), the outline of the superstructure
 528 is superposed to a snapshot taken from our large-domain DNS at $Re_\tau = 1000$ (C1000_L, see
 529 table 1) in figure 12 (d). Considering the channel half-height as the equivalent boundary
 530 layer thickness, it can be seen that the instantaneous structures of the channel form a large-
 531 scale structure of the same height and shape of that found in the boundary layer. Comparing
 532 the frames of figure 12, one can conclude that the unstable mode is reminiscent of the
 533 superstructures.

534 To further illustrate the structure of the unstable eigenmode, its three-dimensional shape,
 535 in the DNS frame of reference, is shown in figure 13 (a) for $Re_\tau = 590$. The modes at
 536 other Reynolds number are totally equivalent, as can be deduced by the various panels of

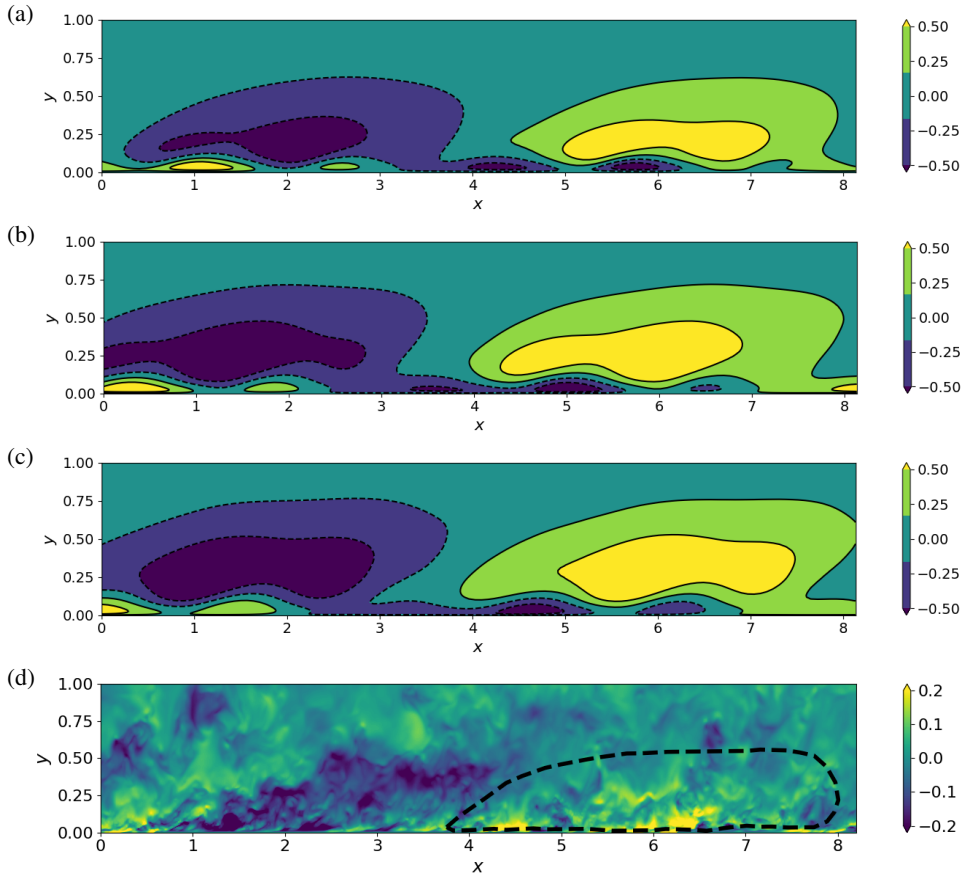


Figure 12: (a-c) Streamwise component of the leading unstable eigenmode reported in the DNS $x - y$ reference frame for (a) $Re_\tau = 590$, (b) $Re_\tau = 1000$ and (c) $Re_\tau = 2000$ ($\alpha = 1.0$, $A_s = 0.20$, $\theta \approx 14^\circ$, $N_u = 60$). The amplitude of the mode is normalized to one. (d) Instantaneous streamwise component of the turbulent fluctuation extracted from a snapshot of the large-domain DNS at $Re_\tau = 1000$ (C1000 $_L$, see table 1). The dashed line is a conceptual representation of a superstructure (from figure 13 of Deshpande *et al.* (2023)).

537 figure 12. In a wall-parallel plane, the mode has the shape of a traveling wave. Therefore,
 538 the only meaningful information in this plane are the streamwise and spanwise wavelengths,
 539 already compared to DNS results in figure 10. To improve the comparison of the wall-normal
 540 structure of the leading unstable mode with DNS results, spanwise-wall-normal cuts of the
 541 two are included in figure 13 (panels b and c). Like the base streaks, the resulting modes
 542 also have infinite length along a given axis. However, they have larger dimensions in the
 543 cross-plane. Therefore, they effectively represent structures larger than the base streaks.

544 The panels of figures 12 and 13 show that a sign change is present in the mode at $y/h \approx 0.1$.
 545 It can be noted, especially in the wall-normal-spanwise plane (figure 13 (c)), that similar
 546 sign changes can be observed at some locations in the DNS snapshots. However, it is not
 547 present a net phase shift as it seems to be present in the unstable mode. One should take
 548 into account that the eigenmode is a linear, therefore idealized, structure, whereas the actual
 549 structures found in the DNS are the result of nonlinear interactions between several modes.
 550 In particular, the nonlinear development of the mode may alter the relative amplitude of the

(a)

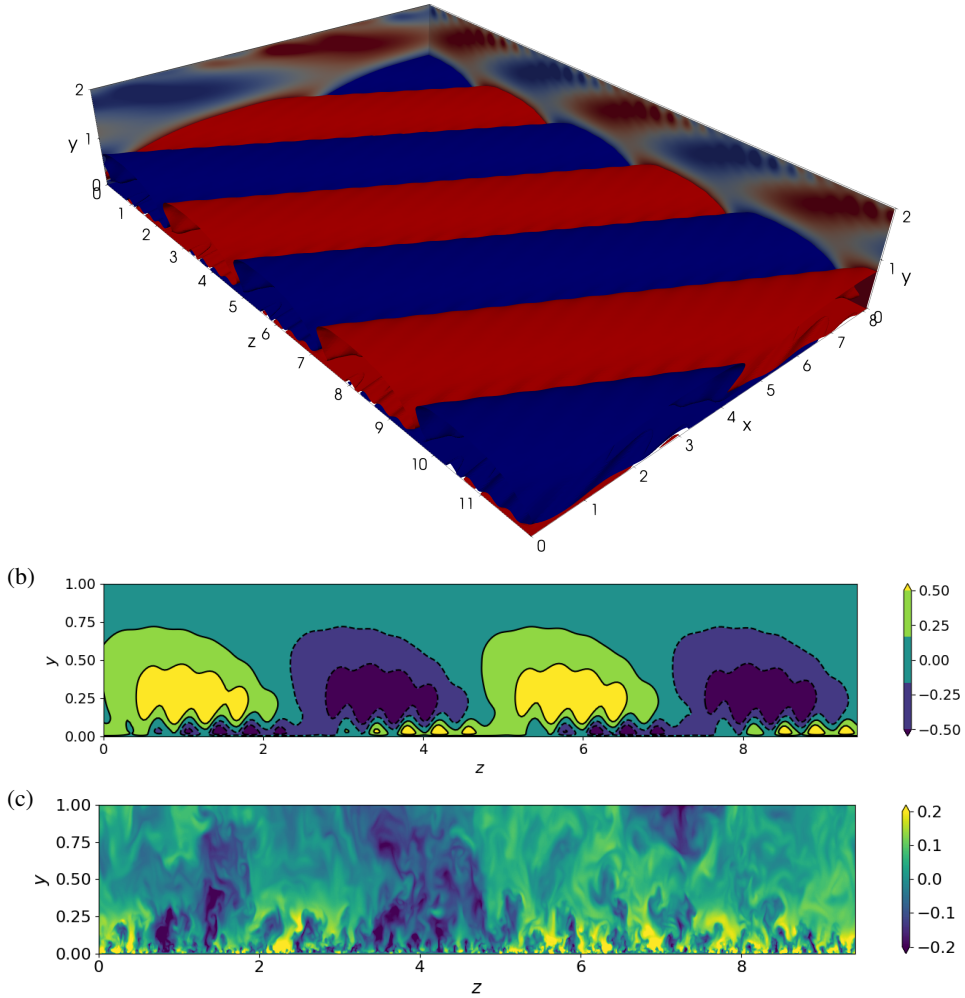


Figure 13: (a) Streamwise velocity component iso-contours (mode amplitude normalized to one, contour value 0.1) of the leading unstable eigenmode reported in the DNS $x-y-z$ reference frame at $Re_\tau = 1000$ ($\alpha = 1.0$, $A_s = 0.20$, $\theta \approx 14^\circ$, $N_u = 60$). (b) $y-z$ cut of the same mode in (a). (c) Instantaneous streamwise component of the turbulent fluctuation extracted from a snapshot of the large-domain DNS at $Re_\tau = 1000$ (C1000 $_L$), $y-z$ cut.

551 near-wall small-wavelength wave with respect to the amplitude of the large-wavelength wave
 552 such that the sign of the large-wavelength wave would prevail and mostly remove the sign
 553 change. Unfortunately, such a nonlinear development would transfer energy also to other
 554 wavelengths, such that the problem would become easily not prone to detailed analysis.
 555 Therefore, this feature of the eigenmodes is not currently well understood.

556 4. Conclusion

557 In this work, the detuned stability of turbulent near-wall streaks and its connection to the
 558 appearance of large-scale motions in the flow is investigated. Streaky structures with an

559 arbitrarily small inclination in the streamwise-spanwise plane are extracted by POD from
 560 minimal-flow-unit DNS data. A periodic N_u -array of such structures superposed to the
 561 turbulent mean profile is considered as base flow for the stability analysis. Unresolved
 562 turbulent motions were included in the linear operator using an eddy viscosity model, as in
 563 previous works (Park *et al.* 2011; Alizard 2015; Cossu 2022). This method allows the study
 564 of turbulent coherent structures without the need for heavy computations in large domains.

565 However, the employed approach has several limitations:

- 566 • First of all, the considered base flow is not a steady solution of the nonlinear equations,
 567 as would be in a classical stability analysis. **The underlying assumption of frozen base flow is**
 568 **only weakly substantiated (see section §3.1)**. Also, a separation of scales argument is difficult
 569 to advocate in fully turbulent flows like those considered in this work;
- 570 • The current method is limited to two-dimensional base flows, whereas realistic near-wall
 571 streaks can cross in the $x - z$ plane, making the candidate base flows three-dimensional;
- 572 • The block-circulant matrix method requires periodic base flows, whereas realistic near-
 573 wall streaks are inclined of different angles, hence not perfectly periodic;
- 574 • The base flow is constructed with a POD mode computed from a given wavenumber
 575 couple $\{k_x, k_z\}$. This mode is then equipped with an amplitude that takes into account the
 576 energy contained in the whole turbulent spectrum. This is a strong approximation, as it means
 577 that a complex chaotic field is ideally represented by only one Fourier mode;
- 578 • As pointed out by Cossu (2022), there is not universal agreement in the literature on
 579 how to close the linear equations (and whether to employ a closure at all). However, we have
 580 verified that eddy viscosities computed from DNS data are not much different from the Cess
 581 (1958) model employed.

582 These shortcomings are, at least in part, shared by most previous studies on the secondary
 583 instability of streaks (Schoppa & Hussain 2002; Marquillie *et al.* 2011; Hack & Zaki 2014;
 584 Hack & Moin 2018; Park *et al.* 2011; Alizard 2015) and also by various works on the stability
 585 of mean flows (for instance, see Cossu (2022); McKeon & Sharma (2010)). In the present
 586 work, it has not been possible to tackle and overcome them, but they are under current
 587 investigation.

588 Notwithstanding the various approximations made, the results display a certain consistency
 589 with the features of large-scale structures reported in previous studies. The results show
 590 that for sufficiently high friction Reynolds number, near-wall streaks can trigger a large-
 591 scale instability, suggesting a possible origin of large-scale motions (LSMs) in wall-
 592 bounded turbulence. There is little qualitative difference between the unstable modes at
 593 $Re_\tau = [590, 1000, 2000]$ and this is compatible with the fact that large structures scale in
 594 outer units (Hutchins & Marusic (2007a); Cossu & Hwang (2017)). Whereas, the $Re_\tau = 180$
 595 case is always stable except for very large amplitudes of the streaks and small wavelengths
 596 of the instability, corroborating the observation that developed large-scale structures are not
 597 expected at this low Reynolds number.

598 A comparison of the computed unstable eigenmodes with DNS and experimental results
 599 was attempted. It was found that the eigenmodes reproduce some features of turbulent
 600 large-scale motions (LSMs). The streamwise and spanwise wavelengths of the large-scale
 601 modulation are compatible with the DNS spectra of Del Alamo *et al.* (2004) and Hoyas &
 602 Jiménez (2006). They also scale according to a power law that is included between the two
 603 power laws $\lambda_z \propto \lambda_x^{0.5}$ and $\lambda_z \propto \lambda_x$ reported by Del Alamo *et al.* (2004). In addition, the
 604 scaling of the wall-normal position of the spectrum inner peak with Re_τ agrees reasonably
 605 well with experimental findings (Vincenti *et al.* 2013; Vallikivi *et al.* 2015), with a slight
 606 improvement with respect to previous linear computations (Hwang 2016). Moreover, it was
 607 found that these traits of the eigenmodes are rather robust with respect to the variation of
 608 the base streaks wavenumber couple $\{k_x, k_z\}$. This consistency sustains at least in part the

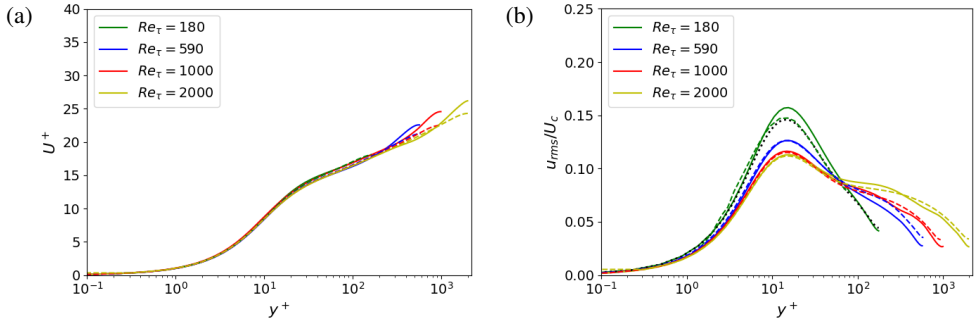


Figure 14: Validation of DNS data. (a) Mean flow scaled in wall units; (b) root mean square of the streamwise velocity fluctuation scaled with the centerline velocity. Dashed lines are reference data: respectively Kim *et al.* (1987), Moser *et al.* (1999), Del Alamo *et al.* (2004) and Hoyas & Jiménez (2006) for $Re_\tau = 180$, $Re_\tau = 590$, $Re_\tau = 1000$ and $Re_\tau = 2000$. The black dotted line is from a DNS at $Re_\tau = 180$ in the domain of Kim *et al.* (1987).

609 applicability of the assumptions made, since it shows that the choice of a given wavenumber
 610 couple is not critical as long as it respects some physical constraints (moderate inclination
 611 of the streaks with respect to the streamwise direction).

612 As the employed base flow, the eigenmode itself is an idealized structure, as it resembles
 613 an oblique traveling wave. Nevertheless, it contains interesting elements. The shape of the
 614 structure in the streamwise/wall-normal plane conceptually recalls the structures found in
 615 recent experimental findings (Deshpande *et al.* 2023) and in the large domain DNS performed
 616 in this study. Similar considerations apply to the structure in the spanwise/wall-normal plane.
 617 Even if in the comparison there are unclear elements (e.g. a near-wall phase shift not well
 618 understood at the moment), which are probably due to the linear nature of the eigenmode,
 619 it can be concluded that the instability returns the correct streamwise/spanwise wavelengths
 620 and a fair wall-normal structure.

621 Therefore, this study brings numerical evidence that the large-scale motions (LSMs)
 622 found in numerical simulations and experiments of wall-turbulence may be the result of an
 623 instability of near-wall structures. Despite the considered Reynolds numbers are still too low
 624 to assess a connection with very-large-scale structures (VLSMs), this work sheds a new light
 625 on the role of instabilities in the dynamics of wall-bounded turbulent flows and the possible
 626 bottom-up scale interaction originated by them.

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 628 matrix method. This work was granted access to the HPC resources of TGCC under the allocation 2023-
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630 **Declaration of interests.** The authors report no conflict of interest.

631 Appendix A. DNS validation

632 DNS computations were performed in minimal flow units in order to extract the base flow
 633 streaks (see table 1). In order to validate the data, velocity statistics are compared to literature
 634 results (Kim *et al.* 1987; Moser *et al.* 1999; Del Alamo *et al.* 2004; Hoyas & Jiménez 2006)
 635 in figure 14. It can be seen that the mean flow and the streamwise fluctuation r.m.s. are
 636 well computed, with minor differences imputable to the use of very small domains. Indeed,
 637 repeating the computation at $Re_\tau = 180$ in a larger domain (see Kim *et al.* (1987)), the
 638 discrepancy in the peak of the r.m.s. in figure 14 (b) disappears. However, it must be noted

639 that these small-size effects influence the flow only for $y \gtrsim L_z/3 \approx 0.25h$ (Flores & Jiménez
640 2010), so the considered minimal flow units are appropriate to extract near-wall coherent
641 structures.

642 **Appendix B. Derivation of linearised perturbation equations**

643 Let us consider a flow field decomposed as follows:

$$644 \quad u_i = U_i + u'_i; \quad p = P + p'. \quad (\text{B } 1)$$

645 The continuity equation implies that, assuming u_i and U_i to be divergence free, then u'_i must
646 also be divergence free. The total flow field verifies the Navier-Stokes momentum equation:

$$647 \quad \frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial \bar{x}_j} = -\frac{\partial p}{\partial \bar{x}_i} + \nu \frac{\partial^2 u_i}{\partial \bar{x}_j^2}, \quad (\text{B } 2)$$

648 which, by introducing the decomposition (B 1), reads:

$$649 \quad \frac{\partial U_i}{\partial t} + \frac{\partial u'_i}{\partial t} + U_j \frac{\partial U_i}{\partial \bar{x}_j} + U_j \frac{\partial u'_i}{\partial \bar{x}_j} + u'_j \frac{\partial U_i}{\partial \bar{x}_j} + u'_j \frac{\partial u'_i}{\partial \bar{x}_j} = -\frac{\partial P}{\partial \bar{x}_i} - \frac{\partial p'}{\partial \bar{x}_i} + \nu \frac{\partial^2 U_i}{\partial \bar{x}_j^2} + \nu \frac{\partial^2 u'_i}{\partial \bar{x}_j^2}. \quad (\text{B } 3)$$

650 One could assume that the base flow U_i also verifies the *unsteady* momentum equation:

$$651 \quad \frac{\partial U_i}{\partial t} + U_j \frac{\partial U_i}{\partial \bar{x}_j} = -\frac{\partial P}{\partial \bar{x}_i} + \nu \frac{\partial^2 U_i}{\partial \bar{x}_j^2}. \quad (\text{B } 4)$$

652 If this is the case, without any further assumption, by subtracting (B 4) from (B 3), the
653 perturbation equation is obtained:

$$654 \quad \frac{\partial u'_i}{\partial t} + U_j \frac{\partial u'_i}{\partial \bar{x}_j} + u'_j \frac{\partial U_i}{\partial \bar{x}_j} + u'_j \frac{\partial u'_i}{\partial \bar{x}_j} = -\frac{\partial p'}{\partial \bar{x}_i} + \nu \frac{\partial^2 u'_i}{\partial \bar{x}_j^2}. \quad (\text{B } 5)$$

655 Then, the linearized equation is obtained by assuming $u'_i \sim \varepsilon \ll 1$, such that the quadratic
656 term $u'_j \partial u'_i / \partial \bar{x}_j$ can be neglected:

$$657 \quad \frac{\partial u'_i}{\partial t} + U_j \frac{\partial u'_i}{\partial \bar{x}_j} + u'_j \frac{\partial U_i}{\partial \bar{x}_j} = -\frac{\partial p'}{\partial \bar{x}_i} + \nu \frac{\partial^2 u'_i}{\partial \bar{x}_j^2}. \quad (\text{B } 6)$$

658 The derivation of this equation is not affected by the generic unsteadiness of U_i . Indeed, these
659 equations are used, e.g., for the computation of Lyapunov exponents of turbulent flows, where
660 U_i is a chaotic trajectory (Nikitin 2018; Ishikawa *et al.* 2018). The assumption of frozen base
661 flow is made when the linearized problem is formulated as an eigenvalue problem, because
662 then the time dependency of the linear operator, which contains U_i , is neglected.

663 When one wants to introduce an eddy viscosity in the linear operator, the derivation
664 is analogous, if the eddy viscosity is introduced from the beginning, i.e. in the total flow
665 equation:

$$666 \quad \frac{\partial U_i}{\partial t} + \frac{\partial u'_i}{\partial t} + U_j \frac{\partial U_i}{\partial \bar{x}_j} + U_j \frac{\partial u'_i}{\partial \bar{x}_j} + u'_j \frac{\partial U_i}{\partial \bar{x}_j} + u'_j \frac{\partial u'_i}{\partial \bar{x}_j} = -\frac{\partial P}{\partial \bar{x}_i} - \frac{\partial p'}{\partial \bar{x}_i} +$$

$$\frac{\partial}{\partial \bar{x}_j} \left[(\nu + \nu_t) \left(\frac{\partial U_i}{\partial \bar{x}_j} + \frac{\partial U_j}{\partial \bar{x}_i} \right) \right] + \frac{\partial}{\partial \bar{x}_j} \left[(\nu + \nu_t) \left(\frac{\partial u'_i}{\partial \bar{x}_j} + \frac{\partial u'_j}{\partial \bar{x}_i} \right) \right]. \quad (\text{B } 7)$$

667 In this case, the flow field $U_i + u'_i$ is seen as the coherent part of the turbulent flow, whereas

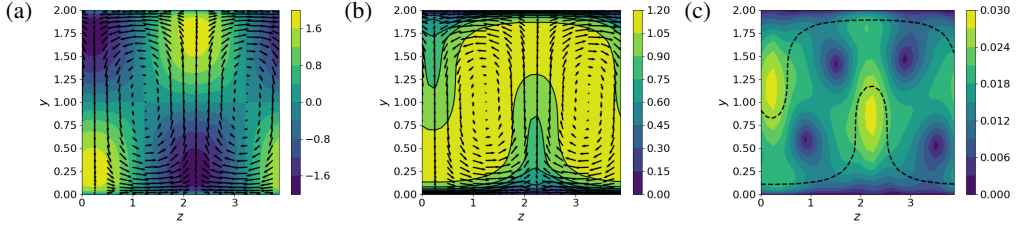


Figure 15: (a): Optimal perturbation for the turbulent mean flow at $Re_\tau = 300$ for $\alpha = 0.0$ and $\beta = \pi/2$. Contours of the streamwise velocity at optimal time and transverse velocity components at initial time. (b): Same as (a) for the optimal perturbation evolved with the nonlinear code. (c): Secondary instability mode for $\alpha = 1.3$: contours of the absolute value of the spanwise velocity component. The dashed line is the critical layer, i.e. where the local base flow velocity is equal to the phase velocity of the unstable mode.

668 the incoherent fluctuations are modelled by the eddy viscosity (Cossu 2022). The base flow
669 is assumed to verify the following *unsteady* equation:

$$670 \quad \frac{\partial U_i}{\partial t} + U_j \frac{\partial U_i}{\partial \bar{x}_j} = -\frac{\partial P}{\partial \bar{x}_i} + \frac{\partial}{\partial \bar{x}_j} \left[(v + \nu_t) \left(\frac{\partial U_i}{\partial \bar{x}_j} + \frac{\partial U_j}{\partial \bar{x}_i} \right) \right]. \quad (\text{B } 8)$$

671 By subtracting (B 8) from (B 7) and assuming small perturbations, one obtains the linearized
672 perturbation equation:

$$673 \quad \frac{\partial u'_i}{\partial t} + U_j \frac{\partial u'_i}{\partial \bar{x}_j} + u'_j \frac{\partial U_i}{\partial \bar{x}_j} = -\frac{\partial p'}{\partial \bar{x}_i} + \frac{\partial}{\partial \bar{x}_j} \left[(v + \nu_t) \left(\frac{\partial u'_i}{\partial \bar{x}_j} + \frac{\partial u'_j}{\partial \bar{x}_i} \right) \right], \quad (\text{B } 9)$$

674 which is equation (2.6) of the main text. We note that the perturbation equation contains an
675 eddy diffusivity term proportional to the perturbation, that can be interpreted as a perturbation
676 of the Reynolds stresses. Again, the frozen base flow assumption comes in when one considers
677 this linearized equation as an eigenvalue problem.

678 Appendix C. Stability code validation

679 The two-dimensional stability code is validated against the results of Park *et al.* (2011) at
680 $Re_\tau = 300$. Their approach is followed: a primary optimal disturbance (figure 15 (a)) is
681 computed from the mean flow transient growth in agreement with Pujals *et al.* (2009); the
682 optimal initial vortices are rescaled with an amplitude $A_v = 0.1$ and evolved nonlinearly;
683 then, the resulting streaks with amplitude $A_s = 0.23$ (figure 15 (b)) are added to the mean
684 flow to form the base flow. Performing a 2D stability analysis with $\alpha = 1.3$, this base flow
685 is found to be unstable. The unstable eigenmode is shown in figure 15 (c) along with the
686 critical layer: it is equivalent to figure 15 (d) in Alizard (2015). The growth rate and the
687 phase velocity of this unstable mode are respectively $\sigma_\tau = 0.021$ and $c/U_c = 0.86$ which are
688 in excellent agreement with Park *et al.* (2011) and Alizard (2015).

689 The block-circulant matrix method was validated in a previous study (Jouin *et al.* 2024).

690 Appendix D. Dependence on the number of sub-units

691 The dependence of the stability results (eigenvalues and eigenmodes) on the number of
692 coupled sub-units N_u is assessed here at $Re_\tau = 2000$. The variation of the leading growth
693 rate is displayed in figure 16 (a) for $\alpha = 0.5$ and $\alpha = 1.0$. We report a variation in the
694 leading growth rate between $N_u = 60$ and $N_u = 90$ of the 2.4% for $\alpha = 0.5$ and of the

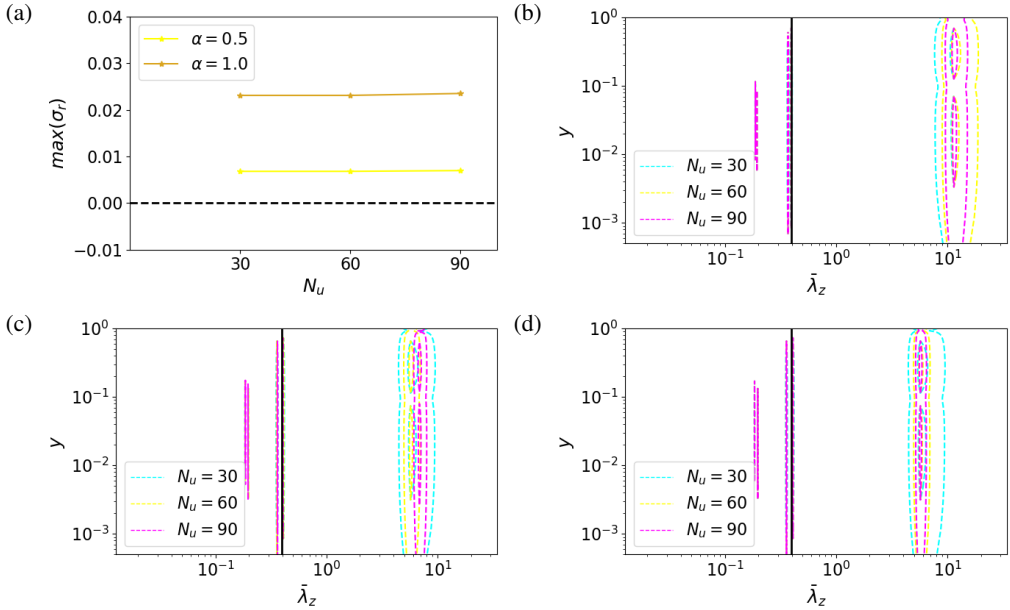


Figure 16: (a): Growth rate of the leading unstable mode (\star) as a function of the number of sub-units N_u for two different values of the α wavenumber and $Re_\tau = 2000$, $k_z = 2$, $A_s = 0.20$. (b-d): Spectral energy content along the \bar{z} direction of the streamwise component of the leading unstable eigenmodes (E_{uu}) for (b) $\alpha = 0.5$ and (c,d) $\alpha = 1.0$. The vertical black line denotes the width of a sub-unit. The eigenmode is normalized to have kinetic energy equal to one. The contour levels are for $E_{uu} = [10^{-4}, 10^{-1}]$. The difference between (c) and (d) is that a different root unity index j is selected for the $N_u = 90$ mode (see appendix D text for a comment).

695 1.8% for $\alpha = 1.0$. The figure clearly shows that this variation is negligible with respect to
 696 the variation due to e.g. the wavenumber α . Therefore, N_u has a secondary influence on the
 697 eigenvalues. Concerning the influence on the eigenmodes, figure 16 (b-d) shows the spectral
 698 energy content in \bar{z} of the leading modes for $\alpha = 0.5$ (b) and $\alpha = 1.0$ (c-d), in the same
 699 spirit of figure 8. It can be seen that the structure of the eigenmodes is not influenced by
 700 N_u . The only difference that may be recovered, depending on the case, is illustrated by the
 701 two panels (c) and (d). The leading mode obtained with $N_u = 60$ corresponds to a root of
 702 unity index (see section 2) $j = 56$, whereas with $N_u = 90$ the leading mode corresponds
 703 to $j = 85$. These unstable modes are compared in panel (c). Whereas, in panel (d) we plot
 704 the $N_u = 90$ unstable mode corresponding to $j = 84$ instead of 85 (together with leading
 705 modes obtained from $N_u = 30$ and 60). We see that the $j = 84$ mode matches well those
 706 obtained with $N_u = 30$ and 60 (indeed $84/56 = 1.5 = 90/60$), while the truly most unstable
 707 mode at $N_u = 90$ ($j = 85$) has a slightly different prominent wavelength. Hence, this slight
 708 shift in the prominent wavelength is what may happen to the most unstable eigenmode when
 709 increasing the total number of sub-units. However, this does not alter the conclusions of the
 710 paper because the prominent wavelengths remain in the neighbourhood of those shown in
 711 figure 10.

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