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A novel method for determining the feasible integral self-stress states for tensegrity structures

3 Aguinaldo Fraddosio¹, Gaetano Pavone^{1,*}, Mario Daniele Piccioni¹

¹Department of Civil Engineering and Architecture (DICAR), Polytechnic University of Bari, Bari,
Italy

6 *E-mail: aguinaldo.fraddosio@poliba.it, mariodaniele.piccioni@poliba.it*

7 *E-mail corresponding author: gaetano.pavone@poliba.it*

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9 Abstract

10 The form-finding analysis is a crucial step for determining the stable self-equilibrated states for 11 tensegrity structures, in the absence of external loads. This form-finding problem leads to the 12 evaluation of both the self-stress in the elements and the shape of the tensegrity structure. This paper 13 presents a novel method for determining feasible integral self-stress states for tensegrity structures, that 14 is self-equilibrated states consistent with the unilateral behaviour of the elements, struts in compression 15 and cables in tension, and with the symmetry properties of the structure. In particular, once defined the 16 connectivity between the elements and the nodal coordinates, the feasible self-stress states are 17 determined by suitably investigating the Distributed Static Indeterminacy (DSI). The proposed method 18 allows for obtaining feasible integral self-stress solutions by a unique Singular Value Decomposition 19 (SVD) of the equilibrium matrix, whereas other approaches in the literature require two SVD. 20 Moreover, the proposed approach allows for effectively determining the Force Denstiy matrix, whose 21 properties are strictly related to the super-stability of the tensegrity structures. Three tensegrity

- structures were studied in order to assess and discuss the efficiency and accuracy of the proposed
- 23 innovative method.
- 24 Paper included in the Special Issue entitled: "Shell and Spatial Structures: Between New Developments
- and Historical Aspects".

26 Keywords

27 Tensegrity structures, self-equilibrium, feasible self-stress states.

28 1. Introduction

- 29 Tensegrity structures are an intriguing class of reticulated systems and hold promising possibilities in
- 30 different applications: from architecture [1,2] to civil engineering [3–6], from biology [7,8] to
- 31 aerospace [9–11], as well as from robotics [12–15] to the design of metamaterials [16–20].
- 32 Originally proposed by Buckminster Fuller [21], tensegrity structures can be defined as a, usually free-
- 33 standing, pre-stressed, pin-jointed system, composed by a network of tensile elements (cables) within a
- 34 discontinuous set of compressed elements (struts). The initial pre-stressed condition allows for the
- rigidity and the stability of the tensegrity structures [22].
- 36 It is evident that the mechanical behaviour of these structures is highly dependent on the self-stress
- 37 states [23]. Thus a complete analysis of tensegrity structures is made of two key points: first, the form-

38 finding problem, and then the study of the response to the external loads [24].

- The process of form-finding depends on the initial input parameters, that is, the geometry of the structure and the level of the self-stress in the elements [25,26]. Commonly, both the geometry and the self-stress are unknown variables of the problem. If only the latter is known, i.e. the internal forces in the elements in the self-equilibrium state are defined, the problem reduces to the seeking of the nodal
- 43 coordinates of the structure, which can be determined from the analysis of the equilibrium states. On

the other hand, if the geometry of the tensegrity structure is known, that is, the nodal coordinates and
the connectivity between elements are prescribed, the problem turns out to be the initial self-stress
identification (*force-finding* problem) [27].

In the latter case, however, difficulties arise with the evaluation of the level of the self-stress and then of suitable self-stress vectors which taking into account both the unilateral behaviour of the elements and the self-equilibrium of the tensegrity structure [28]. This happens, especially, for tensegrity structures with multiple independent self-stress states [29]. Indeed, in general, the independent selfstress modes obtained from the null-space of the equilibrium matrix do not meet the predefined unilateral behaviour of the elements [30]. Thus, it is necessary to determine a special combination of such independent self-stress modes in order to define possible feasible self-stress states [31].

It is worth to recall that, a *feasible self-stress state* is a self-stress state consistent both with the selfequilibrium of the tensegrity structure and the unilateral behaviour of the elements, that is, cables in tension and struts in compression [27]. If a feasible self-stress state also satisfies the symmetry properties of the structure, it takes the name of *feasible integral self-stress state* [32].

58 In the recent past, various efficient analytical or numerical form-finding methods [33] have been

59 proposed: among the others, Force Density Method (FDM) [34–36], programming method [37–40],

60 dynamics relaxation method [41,42], finite-element method [43,44], optimization-based method [45–

61 47].

In the present work, the FDM has been used in order to tackle the self-equilibrium problem fortensegrity structures.

64 The concept of the force density, originally proposed in [48], corresponds to the ratio between the 65 internal forces in the elements and their lengths. Such quantities are clearly affected by the sign, i.e.

positive for cables and negative for struts. By considering the force densities of the elements, the nonlinear problem of the equilibrium can be neatly linearized [49].

Many researchers have made considerable efforts for improving the application of the FDM to the form-finding of tensegrity structures. Among them, Xian et al. [50] proposed an optimization approach based on the FDM and the mixed-integer nonlinear programming for the design of tensegrity structures. The member connectivity, as well as the nodal coordinates and force densities, are simultaneously used as design variables.

Cai et al. [51] studied the form-finding problem of tensegrity structures with multiple equilibrium
modes by means of an equivalent optimization problem of an energy-based objective function with
Lagrange multipliers. Different structural modes corresponding to different symmetry grouping
conditions were achieved.

Also, Cai and Feng [52] proposed an efficient form-finding method based on the optimization method;
here, the force densities of the elements of a tensegrity structure are obtained by minimizing a special
objective function, which satisfies the non-degeneracy necessary condition for the force density matrix.

80 Zhang and Ohsaki [34] presented a numerical method for the form-finding of tensegrity structures. In 81 particular, eigenvalue analysis and spectral decomposition were carried out iteratively to find the 82 feasible set of force densities that satisfies the requirement on the rank deficiency of the equilibrium 83 matrix with respect to the nodal coordinates.

In addition, Zhang et al. [25] presented a highly efficient form-finding method for tensegrity systems based on the structural stiffness matrix defined as the derivative of the out-of-balance force vector with respect to the nodal coordinate vector.

87	Lee et al [53] have studied the truncated polyhedral tensegrity structures by means of a generalized
88	form-finding procedure by using the FDM combined with a genetic algorithm. Additionally, Gan et al.
89	[54] suggested a novel and versatile numerical technique for determining a self-stress state in a
90	combination with a genetic algorithm as a form-finding procedure for an irregular tensegrity structure.
91	Yuan et al. [55] presented a novel and versatile form-finding method for tensegrity structures based on
92	nonlinear equilibrium equations where the nodal coordinates vectors are variables. The input parameters
93	for the form-finding method are the topology, the initial configuration of the structure, the rest lengths,
94	and the axial stiffness of elements.
95	Koohestani [56] utilized the Faddeev-LeVerrier algorithm to generate relationships between force
96	densities of elements, providing explicit analytical conditions for self-stressed states. This method only
97	requires sum and multiplications as major computational operations and overcomes complicated
98	triangular factorizations and eigenvalue decompositions of the symbolic force density matrix.
99	Moreover, Gomez Estrada et al. [57] proposed a numerical form-finding procedure which only requires
100	the specification of the type of each member, i.e. cable or strut, and the connectivity of the nodes.
101	Iterative adjustment of the member forces are made until the state of self-stress is found.
102	Moreover, for describing the mechanical behaviour of tensegrity structures [58-60], the static and
103	kinematic indeterminacy evaluation can be effectively used as a method for structural identification.
104	For defining the contribution of each element to the total degree of indeterminacy of the structure, also
105	taking into account the influence of the material properties, it can be used the distributed static
106	indeterminacy (DSI) value [61]. Thus, DSI can represent the mechanical behaviour of flexible
107	structures in the primary design. Moreover, in [61] a unified method for the DSI evaluation is
108	proposed, both for kinematically determinate and indeterminate structures. It has been highlighted that

since DSI takes into account symmetry properties, a simple but efficient grouping criterion of the
elements of the structure can be established for improving the efficiency of the force-finding method.

111 Notice that DSI values are related to both geometric and stiffness symmetry properties of the structure;

112 moreover, stiffness symmetry (depending on the axial stiffness of the elements) can be inconsistent

113 with the geometric symmetry (depending only on the position of the elements).

114 The application of DSI suggested in [61] concerns the use of DSI as simple and efficient grouping

115 criterion into a specific Force Density Method called *Double Singular Value Decomposition* (DSVD)

116 [62]. Moreover, DSI values were used as symmetry indicators for generating an initial group

117 classification of the elements of a cable-strut structure for performing a DSVD [63]. However such

initial group clustering of elements only reduces the iteration time of seeking a proper grouping schemefor the DSVD.

120 Moreover, once obtained the self-stress states in the elements by using the proposed approach, it is

121 possible to determine the Force Density matrix [66], whose characteristics are crucial for studying the

self-equilibrium problem and the stability conditions for tensegrity structures [67,68].

123 Many authors studied different kinds of problems related to the form-finding of the tensegrity

124 structures based on the properties of the Force Density matrix.

125 Chen et al. [67] pointed out an improved symmetry method for the analytical form-finding of tensegrity

126 structures based on the group representation theory and the FDM. This approach requires only to

127 specify the symmetry properties and the connectivity of the structure. However, with the increase of the

128 element type, the computational complexity of the determination of the Force Density matrix increases.

- 129 Based on the characteristic polynomial of the symbolic Force Density matrix, a general analytical
- 130 scheme for tensegrity form-finding analysis was proposed by Zhang et al. [68]. Also for this case, the

proposed method requires high computational efforts as the geometrical complexity of the structureincreases

Tran and Lee [69] presented a numerical method for form-finding of tensegrity structures in which the topology and the types of members are the only required information; the eigenvalue decomposition of the Force Density matrix and the single value decomposition of the equilibrium matrix are performed iteratively.

137 Another relevant issue concerns stability conditions. In this case, the Force Density matrix plays a

138 fundamental role in the analysis of the necessary and sufficient conditions for the super-stability, i.e.,

139 the property for a tensegrity structure to be stable irrespectively of the selection of materials and of the

140 level of self-stress in the elements [70]. Indeed, a *d*-dimensional tensegrity structure is said to be super-

stable if the Force Density matrix is positive semi-definite and its rank deficiency is equal to d + 1, and it has a *non-degenerate* geometry in the *d*-dimensional space [70].

143 In the literature, to the best of the Author's knowledge, the force-finding problem for tensegrity with

144 multiple independent self-stress modes has been carried out by using cumbersome approaches:

145 optimization techniques, mixed-integer nonlinear programming strategies, spectral decompositions,

146 stiffness matrix evaluations and numerical iterative procedures.

147 Thus, as mentioned above, a more efficient algorithm for determining the feasible integral self-stress 148 states for tensegrity structures by using the DSI values needs to be investigated. In particular, it should 149 be avoided the second *Singular Value Decomposition* (SVD) for reducing time-consuming inherent the 150 grouping operation.

In this paper, an innovative and efficient method for determining feasible integral self-stress states for
tensegrity structures is proposed by considering the Distributed Static Indeterminacy (DSI) evaluation.

153 The only required initial data are the topology of the structure, i.e. the connectivity relations between154 the elements and their types (cables or struts), and the nodal coordinates.

155 Two advantages of the proposed approach can be remarked. First, a unique (SVD) of the equilibrium 156 matrix has to be carried out for determining the independent self-stress modes, which span the null-157 space of this matrix, then through the DSI evaluation it is possible to determine the feasible self-stress 158 states. To this aim, a linear combination of the independent self-stress modes consistent with the 159 flexibility properties of the elements of the tensegrity structure can be evaluated. From this stems the 160 second advantage consisting in the possibility of obtaining different feasible self-stress states according 161 to the design needs by choosing the material parameters of the elements, that is the Young's modulus 162 and the cross-sectional area.

Such innovative method can be especially useful for the analysis of tensegrity structures with multiple independent self-stress states. Unlike the existing methods in literature [27,64,65], in the proposed approach the combined conditions coming from the stiffness symmetry and the geometry symmetry of the tensegrity structure can be satisfied without using further grouping operations, which usually are inferred from a visual inspection of the structure.

168 Furthermore, it can be noted that the Force Density matrix is strictly related to the connectivity

169 properties of the system, i.e. the relations between the elements of the structure and the nodes, and to

170 the level of the self-stress in the elements.

171 The approach here proposed effectively allows for determining the Force Density matrix and its 172 properties with a low computational cost. It reveals to be useful for all the analysis for the tensegrity 173 structures above recalled: the form-finding analysis, the investigation of the super-stability conditions, 174 and the study of the relations between elements and of the self-stress level according to the actual axial

175 stiffness of the elements.

176	The paper is organized as follows: Section 2 briefly introduces the basic idea of the FDM. In Section 3,
177	the concept of the DSI is recalled and its application to the tensegrity structures is explained. Section 4
178	is devoted to the description of the novel method here proposed. Section 5 recalls the definition of the
179	Force Density matrix and illustrates its formulation according to the proposed approach. Finally, for
180	validating the method several well-known tensegrity structures are studied in Section 6.
181	2. Force Density Method
182	In this Section, we briefly recall the self-equilibrium problem for tensegrity structures. The following
183	assumptions are made:
184	• elements (struts and cables) are rectilinear and connected only at their ends by pin-joints;
185	• nodal coordinates and nodal connectivity are given;
186	• no external loads are applied;
187	• the cross-sectional area A of each element remains unchanged under the pre-stress.
188	We consider a tense grity structure with <i>e</i> elements (s_t struts and <i>c</i> cables, that is, $s_t + c = e$) connected
189	to <i>n</i> nodes ($e < 3n$). Nodal coordinates are expressed in a Cartesian orthogonal reference
190	system $O\{\mathbf{e}_x, \mathbf{e}_y, \mathbf{e}_z\}$ and are collected in three vectors \mathbf{x}, \mathbf{y} and $\mathbf{z} \in \mathbb{R}^n$, respectively.
191	By the Graph Theory [71], member connectivity relations can be expressed by means of the so-called
192	<i>Connectivity matrix</i> $\mathbf{C} \in \mathbb{R}^{exn}$ [36]. In particular, if the member <i>k</i> connects the node <i>i</i> to the node <i>j</i> , then
193	the <i>k</i> -th row of C has only two non-zero entries in the <i>i</i> -th and <i>j</i> -th position $(i \le j)$, which are equal to 1
194	and -1 respectively. Hence:
	$\begin{bmatrix} \mathbf{C} \end{bmatrix}_{k,p} = \begin{cases} +1 & \text{if } p = i \\ -1 & \text{if } p = j \\ 0 & \text{otherwise} \end{cases} k = 1, \dots, e, \ p = 1, \dots, n. $ (1)

Furthermore, the length l_k of the *k*-th member of the structure can be expressed as: 195

$$l_{k} = \sqrt{\left(x_{i} - x_{j}\right)^{2} + \left(y_{i} - y_{j}\right)^{2} + \left(z_{i} - z_{j}\right)^{2}}.$$
(2)

For our purposes, the matrix $L \in \mathbb{R}^{exe}$ is defined as the diagonal matrix by collecting the lengths of the elements.

198 The self-equilibrium problem can be solved by using FDM. To this aim, for the *k*-th element of the 199 structure it is possible to define the force density q_k :

$$q_k = \frac{t_k}{l_k},\tag{3}$$

200 where t_k denotes the internal force in the element k (t_k is positive for cables and negative for struts) in

201 the self-stress state. Force densities of the elements can be grouped in the vector $\mathbf{q} \in \mathbb{R}^{e} = \{q_{1}, q_{2}, ..., e_{n}\}$

202 q_k }, whose matrix diagonalization is $\mathbf{Q} \in \mathbb{R}^{exe}$, i.e., $\mathbf{Q} = diag(\mathbf{q})$.

203 Considering both Eq. (1) and Eq. (3), the equilibrium equations for the tensegrity structure in the three 204 directions \mathbf{e}_x , \mathbf{e}_y , and \mathbf{e}_z can be then expressed in the following matrix linear form [33]:

$$\begin{cases} \mathbf{C}^{\mathrm{T}}\mathbf{Q}\mathbf{C}\mathbf{x} = \mathbf{0} \\ \mathbf{C}^{\mathrm{T}}\mathbf{Q}\mathbf{C}\mathbf{y} = \mathbf{0}, \\ \mathbf{C}^{\mathrm{T}}\mathbf{Q}\mathbf{C}\mathbf{z} = \mathbf{0} \end{cases}$$
(4)

205 where the superscript "T" indicates the usual matrix transposition operation.

Alternatively, by considering the element internal forces vector $\mathbf{t} \in \mathbb{R}^e = \{t_1, t_2, ..., t_k\}$, the equilibrium equations in Eq. (4) can be written as:

$$\begin{cases} \mathbf{C}^{\mathrm{T}} diag(\mathbf{C}\mathbf{x})\mathbf{L}^{-1}\mathbf{t} = \mathbf{0} \\ \mathbf{C}^{\mathrm{T}} diag(\mathbf{C}\mathbf{y})\mathbf{L}^{-1}\mathbf{t} = \mathbf{0}. \\ \mathbf{C}^{\mathrm{T}} diag(\mathbf{C}\mathbf{z})\mathbf{L}^{-1}\mathbf{t} = \mathbf{0} \end{cases}$$
(5)

By introducing the *equilibrium matrix* $\mathbf{A} \in \mathbb{R}^{3nxe}$ [66], Eq. (5) can be rewritten in a compact form as:

$$\mathbf{At} = \mathbf{0},\tag{6}$$

209 where the equilibrium matrix **A** can be expressed as:

$$\mathbf{A} = \begin{bmatrix} \mathbf{C}^{\mathrm{T}} diag(\mathbf{C}\mathbf{x})\mathbf{L}^{-1} \\ \mathbf{C}^{\mathrm{T}} diag(\mathbf{C}\mathbf{y})\mathbf{L}^{-1} \\ \mathbf{C}^{\mathrm{T}} diag(\mathbf{C}\mathbf{z})\mathbf{L}^{-1} \end{bmatrix}.$$
(7)

Let r_A be the rank of **A**; if $r_A < e$, non-trivial solutions exist. These non-trivial solutions correspond to *s independent self-stress modes*, which can be viewed as the bases of the vector space of the internal forces in the elements, with:

$$s = e - r_A \ge 1. \tag{8}$$

Hence, it is possible to define a matrix $\mathbf{S} \in \mathbb{R}^{exs}$ whose *i*-th column is the \mathbf{s}_i independent self-stress mode, i.e.:

$$\mathbf{S} = [\mathbf{s}_{1}, \mathbf{s}_{2}, \dots, \mathbf{s}_{s}] = \begin{bmatrix} s_{11} & s_{21} & \cdots & s_{s1} \\ s_{12} & s_{22} & \cdots & s_{s2} \\ \vdots & \vdots & \cdots & \vdots \\ s_{1e} & s_{2e} & \cdots & s_{se} \end{bmatrix}.$$
(9)

A general solution of Eq. (6) can be determined as a linear combination of *s* independent self-stress
modes [29], that is:

$$\mathbf{t} = \mathbf{S}\boldsymbol{\alpha},\tag{10}$$

where α_i , i = 1, 2, ..., s, are arbitrary real coefficients of the linear combination collected in the vector **a** 218 $\in \mathbb{R}^s$. If the null-space of the equilibrium matrix **A** is spanned by a unique independent self-stress mode, i.e. if s = 1, then such vector represents the only feasible self-stress state of the structure. In this case, the matrix **S** becomes a column vector. It can be noted that in this case, the independent self-stress mode should be consistent with the unilateral behaviour of the elements for determining the feasible selfstress states.

If there are multiple independent self-stress modes, i.e. if s > 1, then it is necessary to calculate suitable linear combinations of these bases by means of Eq. (10) since such modes, usually, do not satisfy the unilateral behaviour of the elements as well as the symmetry of the structure.

227 Indeed, independent self-stress modes resulting from the null-space of the equilibrium matrix usually

only satisfy the nodal equilibrium conditions, thus cannot be utilized directly. On the other hand,

229 unilateral conditions related to the mechanical behaviour of struts and cables are not considered in the

230 formulation of the matrix **A**.

However, for statically indeterminate structures (s > 1) exhibiting symmetry properties, as is often the case for tensegrity structures, many elements can be collected into suitable groups according to the symmetry [28]. In this vein, the evaluation of Eq. (10) can be simplified taking into account the symmetry constraints of the geometry of the structure, that is, the same self-stress can be assigned to elements in the same symmetric position. Thus, it can be viewed as a constraint on the self-stress distribution in the elements of the structure.

Definitively, the aim is the evaluation of the self-stress distribution in the elements consistent with the
symmetry properties of the structure and their unilateral behaviour, that is the feasible integral selfstress states.

240 **3.** Distributed static indeterminacy

Let $\mathbf{d} \in \mathbb{R}^{3n}$, and $\mathbf{e} \in \mathbb{R}^{e}$ denote the vector of infinitesimal nodal displacements and the vector of member elongations, respectively. It is possible to define the relations among such kinematic variables in terms of the *compatibility matrix* $\mathbf{B} \in \mathbb{R}^{ex3n}$ [60] such that:

$$\mathbf{Bd} = \mathbf{e}.$$
 (11)

From the principle of virtual work, it follows that $\mathbf{B} = \mathbf{A}^{T}$ [58]. Let r_{B} be the rank of \mathbf{B} ($r_{B} = r_{A}$); then the number *m* of the possible *mechanisms* which span the null-space of \mathbf{B} is $m = 3n - r_{B}$. Moreover, the number m_{i} of the *infinitesimal mechanisms* can be obtained by excluding the rigid-body motions in the three-dimensional space, i.e., $m_{i} = m - 6$.

Taking into account the effects of initial elongations e_k , k = 1, 2, ..., e, under the pre-stress, and by assembling the initial elongations vector $\mathbf{e}_0 \in \mathbb{R}^e$, constitutive equations for the tensegrity structures can be then expressed as [61]:

$$\mathbf{e} = \mathbf{e}_0 + \mathbf{F} \mathbf{t},\tag{12}$$

where $\mathbf{F} \in \mathbb{R}^{exe}$ is the diagonal *flexibility matrix*, whose *k*-th diagonal entry is l_k/E_kA_k , with E_k and A_k the Young's modulus and the cross-sectional area of the element, respectively.

253 Moreover, in the standard linear algebraic theory of vector spaces, it results that all the information

required for the analysis of a framework are contained in the four fundamental vector spaces associated

with the equilibrium matrix A (for further details about their kinematic and static interpretation [59]).

- 256 In particular, the row-space, the null-space, the column-space and the left null-space of A, can be
- associated with the equilibrium matrix. In particular, the left null-space and the null-space of A are
- 258 spanned by the m_i infinitesimal mechanisms and the *s* independent self-stress modes, respectively. For

what considered below, it is possible to recall the well-known properties of orthogonality among such
vector subspaces [59], thus it is possible to write:

$$\mathbf{S}^{\mathrm{T}}(\mathbf{e}_{0}+\mathbf{F}\mathbf{t})=\mathbf{0},\tag{13}$$

and substituting Eq. (9) in (13):

$$\mathbf{S}^{\mathrm{T}}(\mathbf{e}_{0}+\mathbf{FS}\boldsymbol{\alpha})=\mathbf{0}.$$
(14)

262 It is possible to recall that for a full rank matrix $\mathbf{P} \in \mathbb{R}^{ixj}$, with $j \le i$, the square matrix $\mathbf{P}^T \mathbf{P}$ is always

263 positive definite. Moreover, let $\mathbf{Q} \in \mathbb{R}^{ixi}$ symmetric and positive definite, then $\mathbf{P}^{T}\mathbf{Q}\mathbf{P}$ is a symmetric,

264 non-singular, positive definite matrix. Thus, the matrix $S^{T}FS$ is a symmetric, non-singular, positive 265 definite matrix.

266 Therefore, from Eq. (14) it is possible to determine the vector \boldsymbol{a} as:

$$\boldsymbol{\alpha} = -\left(\mathbf{S}^{\mathrm{T}}\mathbf{F}\mathbf{S}\right)^{-1}\mathbf{S}^{\mathrm{T}}\mathbf{e}_{0}.$$
(15)

267 Hence, the element internal forces vector **t** can be obtained by substituting Eq. (15) into Eq. (10):

$$\mathbf{t} = -\mathbf{S} \left(\mathbf{S}^{\mathrm{T}} \mathbf{F} \mathbf{S} \right)^{-1} \mathbf{S}^{\mathrm{T}} \mathbf{e}_{0}.$$
(16)

By introducing the *diagonal stiffness matrix* $\mathbf{K} \in \mathbb{R}^{exe}$, such that $\mathbf{K} = \mathbf{F}^{-1}$, (the *k*-th diagonal entry of **K** is E_kA_k/*l_k*) Eq. (15) can be rewritten as:

$$\mathbf{t} = -\mathbf{K} \left[\mathbf{FS} \left(\mathbf{S}^{\mathrm{T}} \mathbf{FS} \right)^{-1} \mathbf{S}^{\mathrm{T}} \right] \mathbf{e}_{0} = -\mathbf{K} \mathbf{\Omega} \, \mathbf{e}_{0}, \tag{17}$$

where the square matrix $\Omega \in \mathbb{R}^{exe}$ (=**FS**(**S**^T**FS**)⁻¹**S**^T) correlates different aspects of the structure: the geometrical configuration, the topology and the stiffness properties of the elements, defined by the

272	designers. Since the matrix $S^{T}FS$ in Eq. (15) is always positive definite; Eq. (17) is applicable for both
273	kinematically determinate and indeterminate structures. Equation (17) is a constitutive equation
274	describing the relation between the internal forces in the elements and their initial elongation. From the
275	definition of the square matrix Ω , it results that Ω is an idempotent singular matrix, that is, Ω^2 is equal
276	to Ω , hence its eigenvalues are either 0 or 1. Furthermore, the rank of Ω is equal to the sum of its
277	eigenvalues, or equivalently, is equal to its trace. The sum of all the main diagonal elements γ_i
278	$(i=1,2,\ldots,e)$ is equal, thus, to the total degree s of static indeterminacy of the structure. Such diagonal
279	entries γ_i , collected into the vector $\boldsymbol{\omega} \in \mathbb{R}^e$, are defined in the literature as <i>Distributed Static</i>
280	<i>Indeterminacies</i> (DSI) [63]: indeed γ_i represents the contribution of the <i>i</i> -th element of the structure to
281	its total degree of static indeterminacy.
282	Moreover, it is possible to show that elements having the same symmetry properties have the same DSI
283	values; indeed, DSI can be viewed as an indicator of the symmetry properties of the structure [61].
284	Finally, if the flexibility matrix \mathbf{F} is equal to the identity matrix \mathbf{I} , then the matrix $\boldsymbol{\Omega}$ becomes the
285	matrix $\Omega_m = S(S^TS)^{-1}S^T \in \mathbb{R}^{exe}$, whose diagonal terms can be collected in the vector $\omega_m \in \mathbb{R}^e$. It can be
286	noted that the matrix Ω_m , in addition to the above-recalled algebraic properties of the matrix Ω , is
287	characterized by the further property of being always symmetric. In this particular case, the matrix Ω_m
288	is not affected by the axial stiffness properties of the elements; hence, it is strictly related to the self-
289	equilibrium conditions of the structure.

290 4. The new approach for the determination of the feasible integral self-stress states

We consider a test vector $\mathbf{t}_p \in \mathbb{R}^e$, consistent with the sign of the internal forces in the elements, i.e. 291

292 positive in the cables and negative in the struts, built as follows:

$$\mathbf{t}_{\mathrm{p},i} = \begin{cases} +1 & \text{if element } i \text{ is a cable} \\ -1 & \text{if element } i \text{ is a strut} \end{cases}, \ i = 1, \dots, e.$$
(18)

By considering a single element of the structure subjected to an initial elongation, it results that shortening generates tension, while extension creates compression. Thus, an initial elongations vector e_0 can be associated to the test vector \mathbf{t}_p :

$$\mathbf{e}_0 = -\mathbf{F}\mathbf{t}_p. \tag{19}$$

296 Substituting Eq. (19) into Eq. (17), we have:

$$\mathbf{t} = \mathbf{K} \mathbf{\Omega} \mathbf{F} \mathbf{t}_{\mathbf{p}}.$$

297 It is easy to prove that **K** Ω **F** is equal to Ω ^T (see Appendix A); therefore, Eq. (20) can be rearranged as:

$$\mathbf{t}_{n} = \mathbf{\Omega}^{\mathrm{T}} \mathbf{t}_{\mathrm{p}}.$$
 (21)

298 From the definition of the matrix Ω and from Eq. (18), the internal forces vector \mathbf{t}_n obtained from the 299 Eq. (21) takes into account both the unilateral behaviour of the elements and the self-equilibrium 300 conditions of the structure. Moreover, as it results from the numerical experiments performed in 301 Section 6, symmetric assignments of the axial stiffness of the elements lead to a symmetric distribution 302 of the internal forces in the elements. Thus, such a vector represents a feasible integral self-stress vector 303 for the tensegrity structure. Moreover, it is worth to observe that, since the definition of the matrix Ω , the Eq. (21) is strictly related to the material properties of the elements, represented by the matrix **F**. 304 In order to verify the accuracy of the numerical analyses performed in Section 6, the vector $\varepsilon_u \in \mathbb{R}^e$ 305 represents the unbalanced residual normalized internal forces defined as: 306

$$\boldsymbol{\varepsilon}_u = \mathbf{A} \mathbf{t},\tag{22}$$

307 and its Euclidean norm can be used to evaluate the numerical errors.

- 308 The proposed method, coded using Wolfram Mathematica 11.0, can be outlined as follows. Assigned
- 309 the element connectivity, by means of the matrix **C**, and the geometry of the structure in terms of the
- 310 nodal coordinate vectors **x**, **y** and **z**, then:
- 311 Step 1: Assemble the equilibrium matrix A by using Eq. (7).
- Step 2: Define the material parameters of the elements, that is, the Young's modulus E_k and the crosssectional area A_k , and then construct the flexibility matrix **F**.
- Step 3: Collect the prototype vector \mathbf{t}_p , according to the unilateral behaviour of the elements, by means of Eq. (18).
- 316 Step 4: Determine the null-space of the equilibrium matrix A and then assemble the self-stress matrix317 S.
- 318 Step 5: Calculate the matrix Ω and thereafter evaluate the feasible self-stress states t_n by using the Eq. 319 (21).
- 320 Step 6: Compute the norm of the unbalanced residual normalized internal forces vectors in order to
 321 verify the accuracy of the analyses.
- 322 It is worth to note that for tensegrity structures with a unique independent self-stress mode, that is s = 1,
- 323 the feasible self-stress states calculated by using the Eq. (21) is obviously not affected by the
- 324 assignments of the axial stiffness of the elements. Conversely, for tensegrity structures with multiple
- independent self-stress modes, that is, for s > 1, different choices of the axial stiffness of the elements
- 326 lead to different linear combinations of the above-mentioned independent self-stress modes, hence, to
- 327 different self-stress states.

328 Conclusively, it can be noted that symmetric distributions of the axial stiffness of the elements

329 correspond to symmetric internal forces in the elements, thus lead to feasible integral self-stress states

330 consistent both with the stiffness symmetry and the geometrical symmetry properties of the structure.

5. An efficient approach for determining the Force Density Matrix

In this section, we briefly recall the definition of the Force Density matrix [66], always a square

333 symmetric matrix. In particular, by using the Eq. (4), the equilibrium equations for the tensegrity

334 structure, projected in the three directions \mathbf{e}_x , \mathbf{e}_y , and \mathbf{e}_z , can be then expressed in the following matrix 335 linear form:

$$\begin{cases} \mathbf{D}_{s}\mathbf{x} = \mathbf{0} \\ \mathbf{D}_{s}\mathbf{y} = \mathbf{0}, \\ \mathbf{D}_{s}\mathbf{z} = \mathbf{0} \end{cases}$$
(23)

336 with $\mathbf{D}_{s} \in \mathbb{R}^{exe}$ the Force Density matrix, defined as follows:

$$\mathbf{D}_{s} = \mathbf{C}^{\mathrm{T}} \mathbf{Q} \mathbf{C}.$$

337 A non-degenerate tensegrity structure is super-stable, if the rank deficiency n_D of the Force Density

matrix, that is the number of its null eigenvalues, is equal to d + 1 ($\lambda_1 = ... = \lambda_{d+1} = 0$), and the remaining eigenvalues are strictly positive ($0 < \lambda_{d+2} \le ... \le \lambda_e$).

Eq. (24) represents the standard formulation of the Force Density matrix \mathbf{D}_{s} . The major difficulties in evaluating \mathbf{D}_{s} trough Eq. (24) comes from the determination of the components of the diagonal matrix Q, especially for tensegrity structures with multiple independent self-stress states, as explained in the previous section.

- 344 We recall that independent self-stress modes, usually, are not consistent with the signs of the internal
- 345 forces in the elements, i.e. positive for cables and negative for struts, since the self-equilibrium

conditions do not take into account the unilateral behaviour of the elements. Moreover, it can be
observed that for a tensegrity structure with multiple independent self-stress states the Force Density
matrix obtained by considering a single independent self-stress mode should be indefinite, that is, it is
neither positive semi-definite nor negative semi-definite.

For these reasons, here an alternative formulation of the Force Density matrix \mathbf{D}_{s} is proposed. Indeed, by recalling the algorithm for the determination of the feasible integral self-stress states proposed in the previous section, and by using the Eq. (21) and Eq.(3), it is possible to rewrite \mathbf{D}_{s} as follows:

$$\mathbf{D}_{s} = \mathbf{C}^{\mathrm{T}} \mathbf{Q} \mathbf{C} = \mathbf{C}^{\mathrm{T}} diag \left(\mathbf{L}^{-1} \mathbf{t}_{n} \right) \mathbf{C} = = \mathbf{C}^{\mathrm{T}} diag \left(\mathbf{L}^{-1} \mathbf{\Omega}^{\mathrm{T}} \mathbf{t}_{p} \right) \mathbf{C},$$
(25)

353 where it is recalled that the diagonal matrix **Q** is equal to $diag(\mathbf{L}^{-1}\mathbf{t}_n)$.

Since our approach allows for efficiently evaluating the feasible integral self-stress states of a tensegrity structure (see Sect. 4), it can now effectively employed also for calculating the Force Density matrix that can be determined by performing a unique SVD of the equilibrium matrix of the structure and by evaluating the DSI values of the elements.

Notice that, different choices of the feasible self-stress states lead to different Force Density matricesand, thus, to different eigenvalues.

- Since the geometry and the connectivity properties of the structure are given, the equilibrium matrix A,
 see Eq. (7), remains unchanged, that is, such matrix is constant in the force-finding problem in the Eq.
 (6).
- Thus, the force-density problem in the Eq. (23) can be seen as an alternative representation of the selfequilibrium problem represented by Eq. (7).

365 Hence, rank deficiency, as well as, the sign of non-zero eigenvalues of the Force Density matrix,

366 evaluated by using Eq. (25), remain unvaried irrespective of any feasible self-stress state considered.

367 Hence, by using the Eq. (25) it is possible to effectively evaluate the matrix D_s also for verifying the

368 super-stability conditions for tensegrity structures.

369 6. Numerical examples

370 In this section, three well-known tensegrity structures have been studied in order to compare the results

available in the literature with the results obtained by means of the new method here proposed. In

372 particular, we analyze the following tensegrity structures: the Quadruplex; the Snelson's X beam with

373 three modules; the Octahedral cell.

374 Different assignments of the Young's modulus and the cross-sectional area of the elements were made 375 in order to calculate the corresponding different feasible self-stress states by evaluating the DSI vector 376 $\boldsymbol{\omega}$ and the corresponding internal force vector **t**.

377 Specifically, for each of the three tensegrity structures, the analysis was conducted by considering five378 different conditions:

1) the case in which the flexibility matrix is equal to the identity matrix, that is, $\mathbf{F} = \mathbf{I}$; in this case, the

380 DSI vector coincides with ω_m and the corresponding internal force vector is denoted by \mathbf{t}_{nm} ;

381 2) a possible assignment of the axial stiffness of the elements which lead to the results reported by the

382 literature; in such case, the vector named ω (*literature*) and the vector termed t (*literature*) were

383 calculated

384 3), 4) and 5) two symmetric distribution (called n1 and n2) of the axial stiffness of the elements and a

385 not-symmetric distribution (called n3) of the axial stiffness of the elements, which yield to the

determination of the DSI vectors $\boldsymbol{\omega}_{n1}$, $\boldsymbol{\omega}_{n2}$, $\boldsymbol{\omega}_{n3}$ and of the related internal force vectors \mathbf{t}_{n1} , \mathbf{t}_{n2} , \mathbf{t}_{n3} , respectively.

388 Furthermore, for both the Snelson's X beam with three modules and the Octahedral cell (tensegrity

389 structures with multiple independent self-stress modes) three further distributions of the axial stiffness

of the elements, which also allow for determining results equal to those reported by the literature, havebeen considered.

392 Moreover, for the tensegrity structures analysed, the Force Density matrices have been calculated.

393 Their rank deficiencies, as well as, their eigenvalues have been determined in order to evaluate the

394 super-stability conditions of the structures.

Finally, in order to compare the results corresponding to different stiffness properties, the internal force vectors have been normalized, and the force densities of the elements have been normalized respect to the force density of the elements belonging to the first group.

398 6.1. Quadruplex

The tensegrity Quadruplex analyzed, see Fig. 1, consists of n = 8 nodes and e = 16 elements, i.e. 4 struts and 12 cables, and its geometrical configuration can be found in [61]. In particular, the top square base and the bottom square base are rotated with respect to each other by a twist angle equal to $\pi/4$; such bases are inscribed in a circle of radius equal to 707 mm, the height of the prism is equal to 1000 mm. In Table 1 are shown the assignments of the axial stiffness of the elements of the Quadruplex, whereas the corresponding DSI values of the elements are shown in Fig. 2.



Fig. 1. Quadruplex, perspective view. Thick cylinders represent the struts. Different colours have been assigned according to the value of the internal forces in the elements, which are labelled according to the connectivity matrix

406

Table 1

	$\mathbf{F} = \mathbf{I}$	literature	n1	n2	n3	
Element	$E_{k}A_{k}\left(\mathrm{N} ight)$	$E_kA_k(\mathbf{N})$	$E_{k}A_{k}\left(\mathrm{N}\right)$	$E_{k}A_{k}\left(\mathbf{N}\right)$	$E_{k}A_{k}\left(\mathbf{N} ight)$	Element
struts (1-4)	1645.33	10 ⁶	10 ⁶	10^{6}	106	struts (1-3)
cables (5-12)	1000	$49 \cdot 10^3$	49.10^{3}	$24.5 \cdot 10^3$	$1.5 \cdot 10^{6}$	strut (4)
cables (13-16)	1137.05	$49 \cdot 10^3$	$24.5 \cdot 10^3$	$49 \cdot 10^3$	$24.5 \cdot 10^3$	cables (5-10)
					$49 \cdot 10^3$	cables (11-12)
					$49 \cdot 10^3$	cables (13-15)
					$39.2 \cdot 10^3$	cable (16)

Axial stiffness of the elements of the Quadruplex



Fig. 2. DSI of the elements of the Quadruplex for different assignments of the axial stiffness.

408

409 As it can be observed, by increasing the axial stiffness of an element of the structure, its DSI value 410 decreases whereas the DSI values of the other elements increase. Moreover, symmetric assignments of 411 the axial stiffness, namely the first four cases analyzed, lead to a symmetric distribution of DSI values. 412 The rank of the equilibrium matrix A is 15, thus the structure has one self-stress mode, i.e. s = 1, and it 413 possesses 3 infinitesimal mechanisms. Furthermore, by using Eq. (21), it is possible to evaluate the 414 feasible self-stress states, see Fig. 3. The figure clearly shows that, as expected, the normalized internal 415 force vectors do not change as the axial stiffness's of the elements vary, also for not-symmetric distribution of the axial stiffness. 416

Internal forces in the elements of the Quadruplex



Fig. 3. Internal forces in the elements of the Quadruplex as the axial stiffness of the elements vary 417

418 As it can be noted in Fig. 3, the elements of the Quadruplex can be collected in three groups according

419 to their internal forces, as well as to their force densities: struts (1-4), cables (5-12) and cables (13-16).

420 In particular, the normalized force density of the elements of the first group is equal to -1, the

421 normalized force density of the elements of the second group is equal to about 0.7071, and the

422 normalized force density of the elements of the third group is equal to 1.

423 Moreover, from the analysis of the Force Density matrix, it results that the Quadruplex is a super-stable

424 tensegrity structure. Indeed, its Force Density matrix is a positive semi-definite matrix with four zero

- 425 eigenvalues, as shown in Fig. 4. Obviously, the corresponding eigenvalues evaluated in the five
- 426 different assignments of the axial stiffness of the elements are identical.



Fig. 4. Eigenvalues of the Force Density matrices of the Quadruplex

427

428 Finally, the norms of the unbalanced residual normalized internal forces vectors are calculated in order

429 to verify the accuracy of the method. As it is shown in Table 2, such norms are close to zero.

Table 2

Norm of the unbalanced residual normalized internal forces vectors of the Quadruplex

	\mathbf{t}_{nm}	t (literature)	\mathbf{t}_{n1}	\mathbf{t}_{n2}	\mathbf{t}_{n3}
$[[\boldsymbol{\varepsilon}_{u}]]$	$5.54 \cdot 10^{-16}$	5.73.10-16	5.76.10-16	6.03.10-16	$5.82 \cdot 10^{-16}$

430

431 6.2. Snelson's X beam with three modules

432 The Snelson's X beam shown in Fig. 5 is made of three modules; its topology and geometry are

433 described in [27]. The Snelson's elementary module has dimensions in x and y directions equal to 3000

- 434 mm and 2000 mm, respectively. The tensegrity structure has 8 nodes and it is composed of 16
- 435 elements, 10 cables and 6 struts. From the analysis of the null-space of the equilibrium matrix A, it

436 results that its rank is equal to 13, thus the tensegrity structure has 3 independent self-stress states.

437 Moreover, the number of the infinitesimal mechanisms is equal to 0, that is the Snelson's X tensegrity

438 beam analyzed is kinematically determinate. The normalized feasible self-stress states evaluated by

439 means of the algorithm presented in [32] are displayed in Fig. 5.

440



Fig. 5. Snelson's X beam with three modules, perspective view. Thick cylinders represent the struts. Different colours have been assigned according to the value of the internal forces in the elements, which are labelled according to the connectivity matrix

441

In Table 3 are listed the assignments of the axial stiffness of the elements of the Snelson's X beam,

443 whereas the related DSI values of the elements are shown in Fig. 6.

In order to evaluate how the internal forces vary as the stiffness properties of a single group of the

elements change, the case n1 and the case n2 differs only for the fact that in the case n2 the stiffness

446 assigned to the cables 9-10 is greater than the stiffness assigned to the same elements in the case n1.

447 As well as for Quadruplex, by increasing the axial stiffness of cables 9-10, their DSI values decrease,

448 whereas DSI values of the other elements increase.

Table 3

	$\mathbf{F} = \mathbf{I}$	literature	n1	n2	n3	
Element	$E_{k}A_{k}\left(\mathrm{N} ight)$	$E_{k}A_{k}\left(\mathrm{N} ight)$	$E_{k}A_{k}\left(\mathrm{N} ight)$	$E_{k}A_{k}\left(\mathrm{N} ight)$	$E_{k}A_{k}\left(\mathbf{N}\right)$	Element
cables (1-4)	3000	$9.714 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$9.714 \cdot 10^{6}$	cables (1-2)
cables (5-6)	3000	6.48·10 ⁵	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	cables (3-6)
cables (7-8)	2000	$18.78 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$6.476 \cdot 10^{6}$	cables (7-9)
cables (9-10)	2000	$24 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$16.19 \cdot 10^{6}$	$12.952 \cdot 10^{6}$	cable (10)
struts (11-14)	3605.55	$19.428 \cdot 10^7$	$19.428 \cdot 10^7$	$19.428 \cdot 10^7$	$16.19 \cdot 10^7$	struts (11-12)
struts (15-16)	3605.55	19.428·10 ⁷	19.428·10 ⁷	19.428·10 ⁷	19.428·10 ⁷	struts (13-16)

Axial stiffness of the elements of the Snelson's X beam with three modules

449



DSI of the elements of the Snelson's X beam -3 modules

Fig. 6. DSI of the elements of the Snelson's X beam with three modules for different assignments of the axial stiffness.

451

452 It is worth to note that the feasible self-stress states obtained in literature, see [27], can be obtained for

several distributions of the stiffness of the elements; three examples are listed in Table 4 (named exS1,
exS2, exS3).

Table 4

Axial stiffness of the elements of the Snelson's X beam with three modules which lead to the results obtained in the literature

	exS1	exS2	exS3
Element	$E_{k}A_{k}\left(\mathrm{N} ight)$	$E_k A_k$ (N)	E_kA_k (N)
cables (1-4)	$1.44 \cdot 10^{6}$	$3.238 \cdot 10^4$	$19.428 \cdot 10^{6}$
cables (5-6)	$4.69 \cdot 10^5$	$8.42 \cdot 10^3$	$5.11 \cdot 10^{6}$
cables (7-8)	$64.76 \cdot 10^{6}$	$16.19 \cdot 10^{6}$	$16.19 \cdot 10^{6}$
cables (9-10)	$64.76 \cdot 10^{6}$	$16.19 \cdot 10^{6}$	$3.238 \cdot 10^{6}$
struts (11-14)	$25.9 \cdot 10^7$	$97.14 \cdot 10^{6}$	$19.428 \cdot 10^7$
struts (15-16)	$25.9 \cdot 10^7$	$12.95 \cdot 10^7$	$97.14 \cdot 10^{6}$

455

456 The analyses of the feasible self-stress states obtained by using the proposed method lead to the

457 normalized internal forces in the elements shown in Fig. 7.



Internal forces in the elements of the Snelson's X beam - 3 modules

Fig. 7. Internal forces in the elements of the tensegrity Snelson's X beam with three modules as the axial stiffness of the elements vary (note that the origin of the vertical axes is 0.13)

459

It can be observed that for $\mathbf{F} = \mathbf{I}$, for the case named n1, as well as for the case n2, although the distribution of the axial stiffness of the elements is different, the elements of the tensegrity structure can be collected in the same groups according to the normalized internal forces. This happens because the matrix $\mathbf{\Omega}$ takes into account not only the stiffness symmetry but also the geometric symmetry properties of the structure.

By considering both the case n1 and n2, it can be seen that by increasing the axial stiffness of the cables 9-10, their internal forces increase. Simultaneously, the tensile internal forces in the cables 1-4 and in the cables 7-8, as well as the compressive internal forces in the struts 11-14 decrease. At the same time, the tensile internal forces in the cables 5-6, as well as, the compressive internal forces in the struts 15-16 increase. 470 Such sensitivity analyses can be easily conducted by varying the Young's modulus and the cross-

471 sectional area either of a unique element of the structure or of a single group of the elements.

472 Moreover, the same behaviour can be noted by examining the force densities of the elements, listed in

473 Table 5, normalized respect to the force density of the first group. Such feasible force densities are in

474 perfect agreement with the results obtained in [27] (refer to Table III in the reference).

475

Table 5

Force densities of the elements of the Snelson's X beam with three modules, normalized with respect to the force density of the first group

	$\mathbf{F} = \mathbf{I}$	literature	nl	n2	n3	
Element	q_k	q_k	q_k	q_k	q_k	Element
cables (1-4)	1	1	1	1	1	cables (1-2)
cables (5-6)	0.92	1	0.87	0.92	1.02	cables (3-4)
cables (7-8)	1	1	1	1	0.95	cables (5-6)
cables (9-10)	1.92	2	1.87	1.92	1	cable (7)
struts (11-14)	-1	-1	-1	-1	1.02	cable (8)
struts (15-16)	-0.92	-1	-0.87	-0.92	1.95	cable (9)
					1.97	cable (10)
					-1	struts (11-12)
					-1.02	struts (13-14)
					-0.95	struts (15-16)

The eigenvalues of the Force Density matrices D_s , calculated by using the Eq. (25), are shown in Fig. 8. It can be observed that these matrices are semi-positive definite and their rank deficiencies are equal to 5. However, such Snelson's X beam has a degenerate geometry in a three-dimensional space, thus, it is not super-stable.



Fig. 8. Eigenvalues of the Force Density matrices of the tensegrity Snelson's X beam with three modules

481

482 The norms of the unbalanced residual normalized internal forces vectors $[[\boldsymbol{\varepsilon}_u]]$ are shown in Table 6,

483 and it can be observed that such values are extremely close to 0, which demonstrates the accuracy of

```
the proposed method.
```

Table 6

Norm of the unbalanced residual normalized internal forces vectors of the Snelson's X beam with three modules

$[[\mathbf{\epsilon}_u]]$	$2.54 \cdot 10^{-16}$	$4.74 \cdot 10^{-16}$	$2.35 \cdot 10^{-16}$	$2.15 \cdot 10^{-16}$	$3.04 \cdot 10^{-16}$

486 6.3. Octahedral cell

485

487 The Octahedral cell, shown in Fig. 9, is made of 6 nodes and 15 elements, 12 cables and 3 struts. Its 488 topology and geometry are illustrated in [29,72]. In particular, the length of the vertical strut (strut 15) 489 is equal to 1000 mm, whereas the lengths of the horizontal struts (struts 13-14) are equal to about 490 666.667 mm. The feasible self-stress states presented in the literature [29,72] are also shown in Fig. 9.



Fig. 9. Octahedral cell, perspective view. Thick cylinders represent the struts. Different colours have been assigned according to the value of the internal forces in the elements, which are labelled according to the connectivity matrix

492 The analysis of the equilibrium matrix A conducts to 3 independent self-stress modes, that is s = 3, and

- 493 0 infinitesimal mechanisms, thus the Octahedral cell is a statically indeterminate and kinematically
- 494 determinate tensegrity structure.

495 In Table 7 are listed the distribution of the axial stiffness of the elements of the Octahedral cell,

496 whereas the related DSI values of the elements are illustrated in Fig. 10Fig. 9. In particular, the case n2

497 differs from the case n1 only for the axial stiffness of the cables 1-4.

498

Table 7

Axial stiffness of the elements of the Octahedral cell

	$\mathbf{F} = \mathbf{I}$	literature	n1	n2	n3	
Element	E_kA_k (N)	$E_{k}A_{k}\left(\mathbf{N}\right)$	$E_k A_k$ (N)	$E_{k}A_{k}\left(\mathbf{N}\right)$	$E_{k}A_{k}\left(\mathbf{N}\right)$	Element
cables (1-4)	471.405	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	16.19·10 ⁶	$9.714 \cdot 10^{6}$	cables (1-2)
cables (5-12)	600.925	19.99·10 ⁶	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	$3.238 \cdot 10^{6}$	cables (3-12)
struts (13-14)	666.667	$48.57 \cdot 10^{6}$	$65.94 \cdot 10^{6}$	$65.94 \cdot 10^{6}$	$65.94 \cdot 10^{6}$	strut (13)
strut (15)	1000	$48.57 \cdot 10^{6}$	65.94·10 ⁶	65.94·10 ⁶	$32.97 \cdot 10^{6}$	struts (14-15)





Also for the Octahedral cell it can be observed that by increasing the axial stiffness of the cables 1-4 their DSI values decrease, whereas DSI values of the remaining elements increase. Moreover, also in this case the feasible self-stress states reported in the literature can be obtained for several assignments of the axial stiffness of the elements of the structure. In particular, in Table 8 are listed three possible assignments (termed exO1, exO2, exO3).

506

Table 8

Axial stiffness of the elements of the Octahedral cell which also lead to the results obtained in the literature

exO1	exO2	exO3	

Element	E_kA_k (N)	$E_{k}A_{k}$ (N)	E_kA_k (N)
cables (1-4)	$3.238 \cdot 10^5$	$5.05 \cdot 10^{6}$	$1.619 \cdot 10^{6}$
cables (5-12)	$16.19 \cdot 10^{6}$	$48.57 \cdot 10^{6}$	$43.472 \cdot 10^{6}$
struts (13-14)	$97.14 \cdot 10^{6}$	$12.952 \cdot 10^7$	$19.428 \cdot 10^7$
strut (15)	$40.534 \cdot 10^{6}$	$12.952 \cdot 10^7$	$11.333 \cdot 10^7$

507

As it can be seen in Fig. 11, for **F** equal to the identity matrix, as well as for the cases named n1 and n2, the elements of the Octahedral cell can be collected in the same groups according to the normalized internal forces. In particular, four groups can be identified: cables 1-4, cables 5-12, struts 13-14 and strut 15. Such a grouping scheme is consistent with both the geometrical symmetry and the stiffness symmetry of the structure.

513 Moreover, by comparing the case n1 with the case n2, it emerges that by increasing the axial stiffness

of the cables 1-4 their tensile internal forces decrease, as well as the compressive internal forces in the

515 horizontal struts 13-14 decrease. At the same time, the tensile internal forces in the cables 5-12 and the

516 compressive internal force in the vertical strut 15 increases.



Internal forces in the elements of the Octahedral Cell

Fig. 11. Internal forces in the elements of the Octahedral cell as the axial stiffness of the elements vary (note that the origin of the vertical axes is 0.13)

518

519 The normalized force densities of the elements of the Octahedral cell consistent with the different

520 assignments of the axial stiffness of the members of the structure are listed in Table 9.

521

Table 9

Force densities of the elements of the Octahedral cell, normalized with respect to the force density of

the first group

	$\mathbf{F} = \mathbf{I}$	literature	n1	n2	n3	
Element	q_k	q_k	q_k	q_k	q_k	Element
cables (1-4)	1	1	1	1	1	cables (1-4)

cobles(5, 12)	0.65	0.5	0.76	0.92	0.8	cables
cables (5-12)	0.05	0.5	0.70			(5,7,9,11)
(12,14)	1.65	1.5		1.74	0.01	cables
struts (13-14)	-1.65	-1.5	-1	-1./6	0.81	(6,8,10,12)
strut (15)	-1.3	-1	-0.87	-1.53	-1.8	strut (13)
-					-1.81	strut (14)
					-1.62	strut (15)

522

- 523 By using the Eq. (25) it is possible to determine the Force Density matrices for each of the feasible self-
- 524 stress states; their eigenvalues are shown in Fig. 12.

Eigenvalues of the Force Density matrix of the Octahedral Cell



Fig. 12. Eigenvalues of the Force Density matrices of the tensegrity Octahedral cell

525

526 It can be noted that the Octahedral cell is a super-stable tensegrity structure; indeed, it has a non-

527 degenerate geometry in the three-dimensional space, and its Force Density matrix is semi-positive

528	definite with rank deficiency equal to 4. Such conditions occur for each of the axial stiffness
529	assignments of the elements, thus for each feasible self-stress states obtained by using the proposed
530	approach.
531	Finally, it is possible to calculate the norm of the unbalanced residual normalized internal forces

532 vectors $[[\varepsilon_u]]$, see Table 10, showing the accuracy of the proposed method.

533

Table 10

Norm of the unbalanced residual normalized internal forces vectors of the Octahedral cell

	t _{nm}	t (literature)	\mathbf{t}_{n1}	\mathbf{t}_{n2}	t _n 3
$[[\boldsymbol{\varepsilon}_{u}]]$	6.38·10 ⁻¹⁶	8.05.10-16	6.23·10 ⁻¹⁶	6.76·10 ⁻¹⁶	6.27·10 ⁻¹⁶

534

535 7. Discussion and conclusions

536 A novel efficient method has been proposed for determining feasible self-stress states for tensegrity

537 structures by investigating the Distributed Static Indeterminacy of the tensegrity structure.

538 The proposed methods have some advantages over the existing form-finding methods; in particular: (i)

539 it allows for evaluating feasible self-stress states by performing only a unique Singular Value

540 Decomposition of the equilibrium matrix A; (ii) it only requires, as preliminary information, the

541 connectivity of the elements and their type, i.e. cable or struts, and the nodal coordinates; (iii) it is

- 542 possible to obtain several feasible self-stress states as linear combinations of the independent self-stress
- 543 modes according to the assignments of the axial stiffness of the elements.

This approach becomes particularly efficient for tensegrity structures with multiple self-stress states, as shown in the examined examples since it is not necessary to perform further SVD decompositions or to initialize grouping operations of the elements.

547 Indeed, such approach consists of determining suitable linear combinations of the independent self-548 stress modes according to the axial stiffness of the elements. Thus, it overcomes difficulties arising 549 with complicated optimization techniques, mixed-integer nonlinear programming strategies, spectral 550 decompositions, stiffness matrix evaluations and numerical iterative procedures presented in the 551 literature. This main feature reduces the time consuming of the computational operations.

Moreover, it emerges that different feasible integral self-stress states can be easily obtained. In fact, feasible self-stress states consistent with the symmetry properties of the structures can be simply calculated by considering symmetric axial stiffness assignments to the elements of the tensegrity structure.

From the knowledge of the independent self-stress modes, and the evaluation of the Distributed Static Indeterminacy values related to the axial stiffness of each element, it is possible to address the selfstress identification of the tensegrity structures. Indeed, the proposed procedure bypass the element grouping operations needed in most of the state-of-the-art methods.

Furthermore, since DSI values are indicators that reflect the combined influence of the geometry, topology and axial stiffness of each element, different choices of element's stiffness lead to different feasible self-stress states. In particular, the load-bearing capacity of an element becomes lower as its DSI value increases. Thus, once evaluated the DSI of the elements of the structure, it is possible to calibrate each axial stiffness for achieving the desired mechanical behaviour of the entire structure.
The numerical analyses have shown that the norms of the unbalanced residual internal forces, evaluated for each case, are extremely close to zero, thus the accuracy of the proposed method has been proved.

- 567 Furthermore, the proposed approach allows to effectively determine the Force Density matrix of the
- 568 structure, in order to evaluate the conditions of the super-stability for the tensegrity structure.
- 569 Moreover, the innovative method can be generalized by using a parametric description of the geometry
- 570 of the structures in order to study how the internal forces in the elements vary as the geometrical
- 571 parameters change, also for large-scale tensegrity structures.

572 Appendix A

573 We can explicitly calculate $K\Omega F$ in terms of the matrix S. In particular, we have:

$$\mathbf{K}\mathbf{\Omega}\mathbf{F} = \mathbf{K}\mathbf{F}\mathbf{S}\left(\mathbf{S}^{\mathrm{T}}\mathbf{F}\mathbf{S}\right)^{-1}\mathbf{S}^{\mathrm{T}}\mathbf{F} =$$

= $\mathbf{S}\left(\mathbf{S}^{\mathrm{T}}\mathbf{F}\mathbf{S}\right)^{-1}\mathbf{S}^{\mathrm{T}}\mathbf{F}.$ (A1)

574 Then we evaluate the transpose of the matrix Ω :

$$\boldsymbol{\Omega}^{\mathrm{T}} = \left(\mathbf{FS}\left(\mathbf{S}^{\mathrm{T}}\mathbf{FS}\right)^{-1}\mathbf{S}^{\mathrm{T}}\right)^{\mathrm{T}} = \\ = \left(\left(\mathbf{S}^{\mathrm{T}}\mathbf{FS}\right)^{-1}\mathbf{S}^{\mathrm{T}}\right)^{\mathrm{T}}\left(\mathbf{FS}\right)^{\mathrm{T}} = \mathbf{S}\left(\left(\mathbf{S}^{\mathrm{T}}\mathbf{FS}\right)^{-1}\right)^{\mathrm{T}}\mathbf{S}^{\mathrm{T}}\mathbf{F} = \\ = \mathbf{S}\left(\left(\mathbf{S}^{\mathrm{T}}\mathbf{FS}\right)^{\mathrm{T}}\right)^{-1}\mathbf{S}^{\mathrm{T}}\mathbf{F} = \mathbf{S}\left(\left(\mathbf{FS}\right)^{\mathrm{T}}\mathbf{S}\right)^{-1}\mathbf{S}^{\mathrm{T}}\mathbf{F} = \\ = \mathbf{S}\left(\mathbf{S}^{\mathrm{T}}\mathbf{FS}\right)^{-1}\mathbf{S}^{\mathrm{T}}\mathbf{F}.$$
(A2)

575 Hence, from Eq. (A1) and Eq. (A2) it is clear that **K** Ω **F** is equal to Ω ^T.

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